

### INVESTIGATION OF THE FEASIBILITY OF DEVELOPING LOW PERMEABILITY POLYMERIC FILMS

J. T. Hoggatt

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16. Abstract	<del></del>			
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values as low as $10^{-15}$ cc - mm/c		- mm/cm - sec. cm r	g wirn some	
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program the permeability, flexibil				
bond film were determined at +25°	· · ·		• •	
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process but no additional loss in f	lexibility resulted fro	om the diffusion bondin	g process.	
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#### **FOREWORD**

This report was prepared by The Boeing Company, Aerospace Group, under NASA Contract NAS 3-13326. "Investigation of the Feasibility of Developing Low-Permeability Polymeric Films". The work was sponsored and managed by the National Aeronautics and Space Administration, Lewis Research Center, Materials & Structures Division, Cleveland, Ohio, with Mr. R. F. Lark, Project Manager.

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Performance of this contract was under the direction of the Materials and Processes Organization, Aerospace Group, The Boeing Company. Mr. P. B. Kennedy was Program Manager, and Mr. J. T. Hoggatt, Program Technical Leader. Major participants in the program include:

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#### 1.0 INTRODUCTION AND BACKGROUND

This program was sponsored to investigate the feasibility of a new laminating and coating concept for reducing the gaseous permeability of select polymeric films for cryogenic service.

Polymeric films, namely Mylar and Kapton, have demonstrated good potential as expulsion bladder materials for cryogenic fluid containment (References 1, 2, 3 and 7). One parameter hindering the successful application of these materials in cryogenic bladders is the diffusion of hydrogen and helium through the film causing interply inflation of the bladder. An inflated bladder lowers both fill and expulsion efficiency and in extreme cases renders the bladder completely inoperative. Although interply inflation can result either from leakage or diffusion, permeability tests on unflexed 2-ply material samples have demonstrated interfply inflation can occur as a result of diffusion alone.

Over the years Boeing and others have attempted to reduce the permeability of film by techniques such as vacuum deposition, chemical vapor deposition, electroless plating and laminating metallic foils to or between films. Each of these techniques has been successful to varying degrees in reducing the gaseous permeability, but generally the coating thickness required to provide low permeability resulted in poor flexability at cryogenic temperatures.

In 1969 Boeing began development of a technique for obtaining low permeability film that retains good flexibility characteristics at cryogenic temperatures. The development approach taken at Boeing was to metallize two plies of polymer film and then to diffusion bond the two plies together at the metal-coated surfaces. The thin metallic coating, 5,000A to 10,000A, can be deposited using any convenient process (vacuum deposition, sputtering, electroless plating, etc.). The coating on the basic film may or may not be impermeable to gases, but when the two plies are diffusion bonded together, the resulting metal flow and the statistical distribution of holes in each film seals the pinholes forming a relatively impermeable film-metal-film laminate (Reference 4). The resulting metallic layer is very thin and flexible.

The diffusion bonded laminate concept has several unique advantages which are important in a cryogenic propellant system: (1) permeability is reduced as a result of both metal flow during the diffusion process and the fact that the pinhole in one metal film is covered by the other metal film; (2) the metal coating is at the neutral axis of the "sandwich" so that stressing of the metal in folds during flexing of a bladder will be reduced, thus prolonging integrity; (3) the metal coating will be protected from abrasion damage during subsequent bladder fabrication and assembly; and (4) the metal coating cannot peel or flake during stress, thereby preventing the propellant from becoming contaminated during expulsion. This latter point is an important factor in obtaining a reliable propulsion system. Due to the high processing temperature requirements imposed by the diffusion bonding process, the laminated concept could only be utilized with Kapton film. In the program, the laminated Kapton films were compared to conventional metallized Kapton and Myler films to provide data control.

#### 2.0 SUMMARY

The objective of this program was to determine the feasibility of reducing the gas permeability rate of Mylar and Kapton films without drastically affecting their flexibility characteristics at cryogenic temperatures. This feasibility was established using a concept of diffusion bonding two layers of metallized films together forming a film-metal-film sandwich laminate. The permeability of Kapton film to gaseous helium was reduced from a nominal  $10^{-9}$  cc - mm/cm $^2$ -sec.-cm Hg to  $10^{-13}$  cc - mm/cm $^2$  - sec.-cm Hg with some values as low as  $10^{-15}$  cc - mm/cm $^2$  - sec - cm Hg being obtained. Similar reductions occurred in the liquid hydrogen permeability at -252°C (-423°F). In the course of the program the permeability, flexibility and bond strength of plain, metallized and diffusion bond film were determined at +25°C (+70°F), -195°C (-320°F) and -252°C (-423°F). The flexibility of Kapton film was reduced slightly due to the metallization process, but no additional loss in flexibility resulted from the diffusion bonding process.

Bond strengths of the diffusion bonded laminates were as good as those obtained with the plain Kapton film.

#### 3.0 TEST PROGRAM

The technical phase of this program was divided into three principal tasks. Task I consisted of basic film characterization. Mylar and Kapton film were evaluated for permeability and flexibility to establish a base line for the studies which followed. In Task II ten metallized film concepts were characterized and the three most promising concepts selected for more extensive evaluation in Task III. The details of Task I, II and III are discussed in Sections 3.1, 3.2 and 3.3 respectively.

#### 3.1 Film Characterization

The objective of this task was to characterize each lot of Mylar and Kapton film used in the program. The data then served as a basis for comparison in assessing the effects of metallizing and laminating on reducing permeability. The material selection was limited to two basic materials: type C Mylar film and Kapton polyimide film. Both materials were evaluated in 0.00635 mm (1/4 mil) and 0.0127 mm (1/2 mil) thicknesses. The film selection was limited to these materials since previous programs (References 1-3 and 7) have shown Mylar and Kapton to be the best materials for cryogenic expulsion bladders.

Film characterization consisted of measuring the permeability to helium gas at  $21^{\circ}\text{C}$  (+70°F), -195°C (-320°F) and -252°C (-423°) and to liquid hydrogen at -252°C (-423°F). Determinations were made in both the stressed and unstressed condition. In addition tests for film flexibility bond strength and interply inflation characteristics were made. Details on each of the characterization tests are given below.

#### 3.1.1 Permeability Measurements

As stated above helium permeability determinations were made on each film at  $+21^{\circ}$ C,  $-195^{\circ}$ C and  $-252^{\circ}$ C in both the unstressed and stressed (20% of ultimate strength) conditions. Measurements were also made on the liqud hydrogen permeability rate at  $-252^{\circ}$ C under similar stressed and unstressed conditions.

The test specimen had an overall diameter of 24.13 cm (9.5 inches) with .953 cm (.375 inches) diameter holes on a 10.16 cm (4") radius for mounting bolts. The effective test area was 17.78 cm (7.0") in diameter for an area of 248.16 cm $^2$  (38.47 in $^2$ ). The specimen was installed in the fixture shown in Figure 1 for test.

For the helium permeation tests in the unstressed condition the specimen was pressurized with gaseous helium to a pressure differential ( $\Delta P$ ) of 51.71 mm Hg (1 psi) across the sample. The sample was restrained with a porous backing plate (Figure 1) to maintain a state of zero stress. The cryostat was filled with the appropriate cryogenic (liquid nitrogen for the -195°C tests and liquid hydrogen for the -252°C tests) to maintain the sample at the desired temperature. The test gas passed through coils submerged in the cryostat prior to contacting the specimen to bring the gas to the desired test temperature (see Figure 2).

The entire system was allowed to come to equilibrium prior to making any measurements. This required about 30-45 minutes. The quantity of gas permeating the film sample was then measured with a helium mass spectrometer at standard conditions.

For the gas permeability determination of the film in the stressed condition (20%) the same procedures were utilized except the backing plate was removed from the permeability fixture and the  $\Delta p$  was increased to 103 mm/Hg (1.5 psi) for the 1/2 mil film and held at 51.71 mm Hg (1 psi) for the 1/2 mil film. This in effect stressed the film to 20% of ultimate in the center section. The quantity of pressure to apply to the film was determined by making a plot of material deflection versus pressure (at room temperature) and calculating the stress on the film.

The pressure at which the film stress was approximately 20% of ultimate was then determined by interpolation.

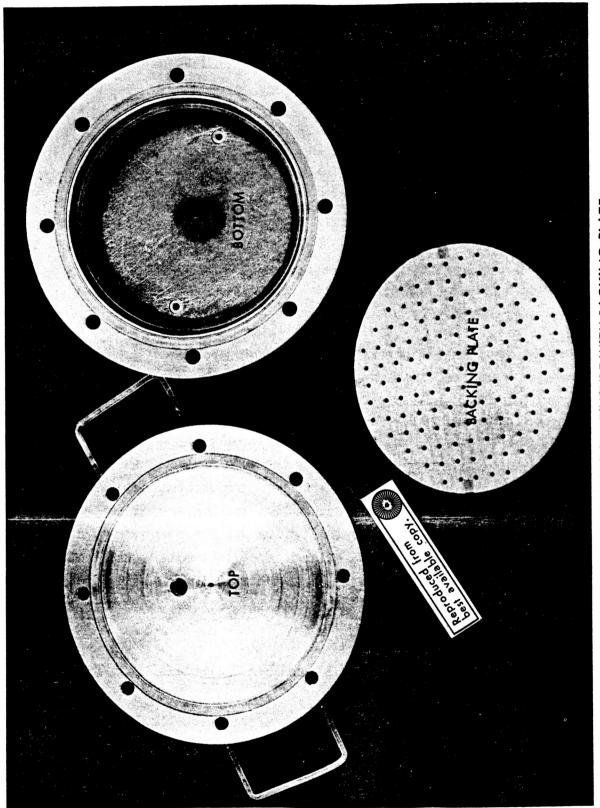


FIGURE 1 PERMEABILITY FIXTURE WITH BACKING PLATE

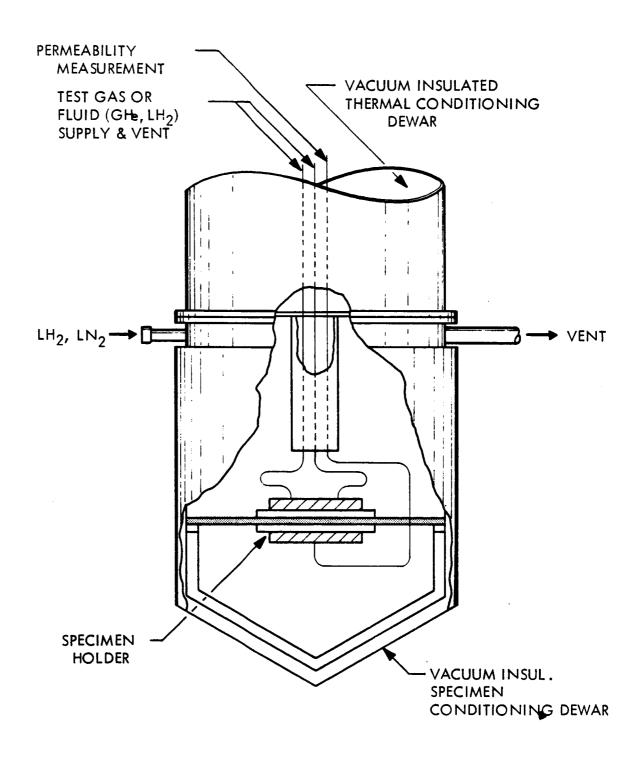


FIGURE 2 PERMEABILITY TEST SYSTEM

Liquid hydrogen permeability rate of the films was determined using the same basic equipment and procedures used for the helium tests. Liquid hydrogen was placed on the inlet side of the sample and in the cryostat. The quantity of hydrogen diffusing through sample was measured with a platinum foil hydrogen analyzer (Boeing built and patented).

As a general procedure to minimize data scatter and provide maximum data correlation, one sample was subjected to all four permeability tests (R. T.,  $-195^{\circ}$ c,  $-252^{\circ}$ c, and LH<sub>2</sub>) without removing it from the test fixture or making any other alterations which may influence the permeability readings. In a few instances, due to specimen leakage, a malfunction or a failure in the seal, a new specimen had to be run. In all cases the system was completely purged after each data point was obtained.

In a few instances the permeability rates exceeded  $10^{-4}$  cc/sec, the capability of the detection equipment, flooding the system. When this occurred the permeability was measured by water displacement using the technique shown in Figure 3. The set-up shown eliminates any back pressure and influences due to water head.

Both the helium mass spectrometer and the hydrogen gauge were calibrated at regular intervals. The helium detector was calibrated with a standard leak (Veco - Model SC-4) and the hydrogen gauge was calibrated with a Faraday cell. The complete test set-up is shown in Figure 4.

The averages of the permeability data are plotted in Figures 5 and 6 and Table I  $_{\mbox{\scriptsize P}}$  Detailed data are contained in Table II and Figures 7 - 16 . In Table I the average permeabilities have been converted to standard units. These data compare favorably with other data reported in the literature and with data reported by duPont in their film brochures.

Approximately twelve attempts were made to conduct permeability readings on stressed 1/4-mil Mylar film at cryogenic temperatures without success. The film ruptured with as little as 10 mm Hg (0.2 psi) pressure differential making it impossible to obtain a reading. By comparison the 1/4-mil Kapton

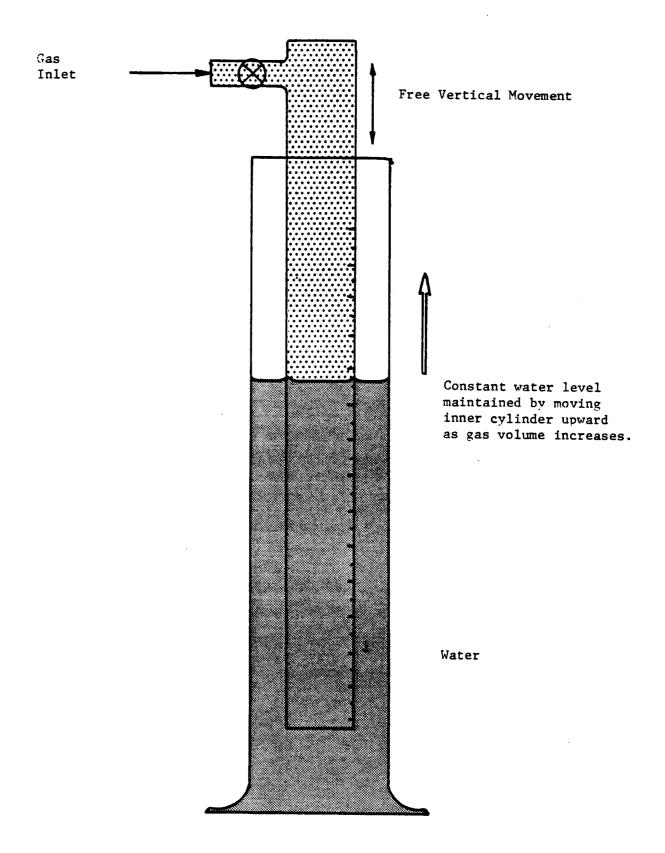


FIGURE 3 GAS VOLUME MEASUREMENT BY WATER DISPLACEMENT

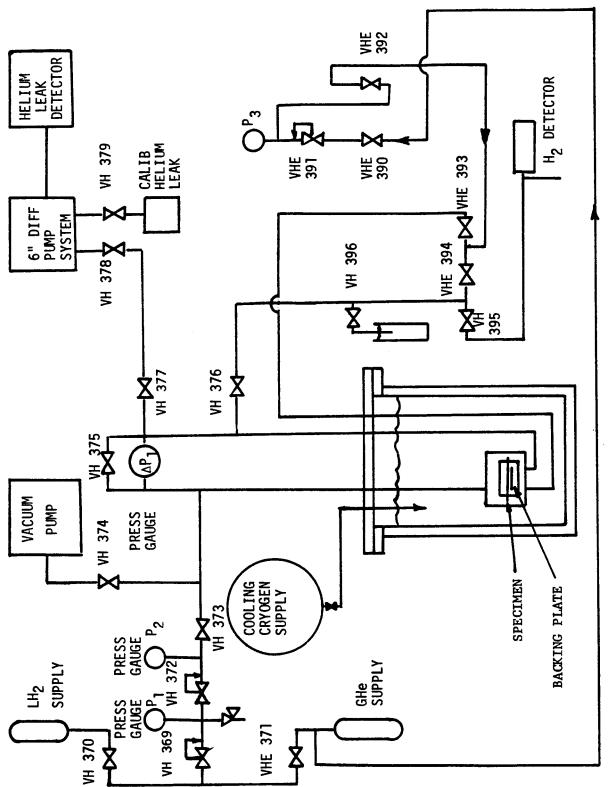


FIGURE 4 Schematic of Permeability Test Set-up

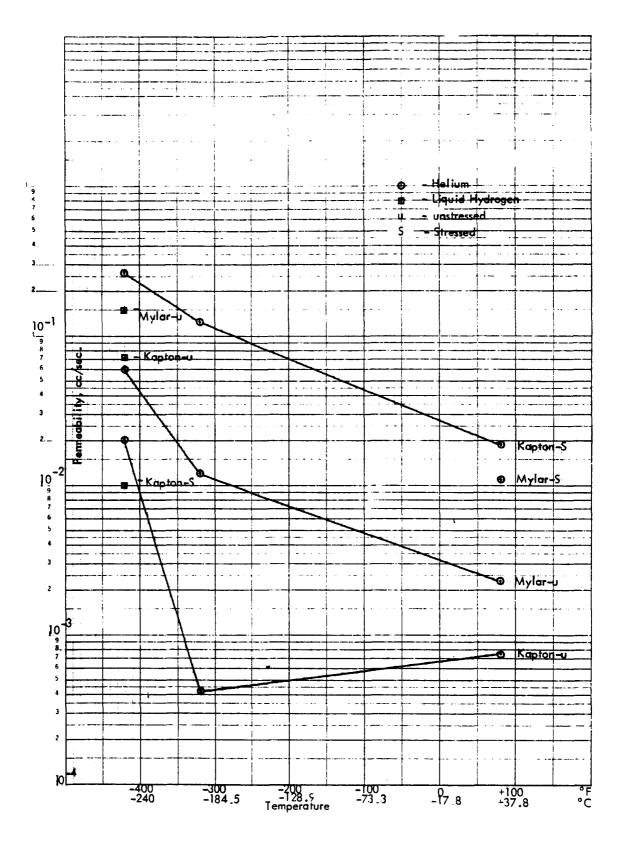


FIGURE 5 - AVERAGE PERMEABILITY - 0.25 MIL (0.00635 MM) FILMS

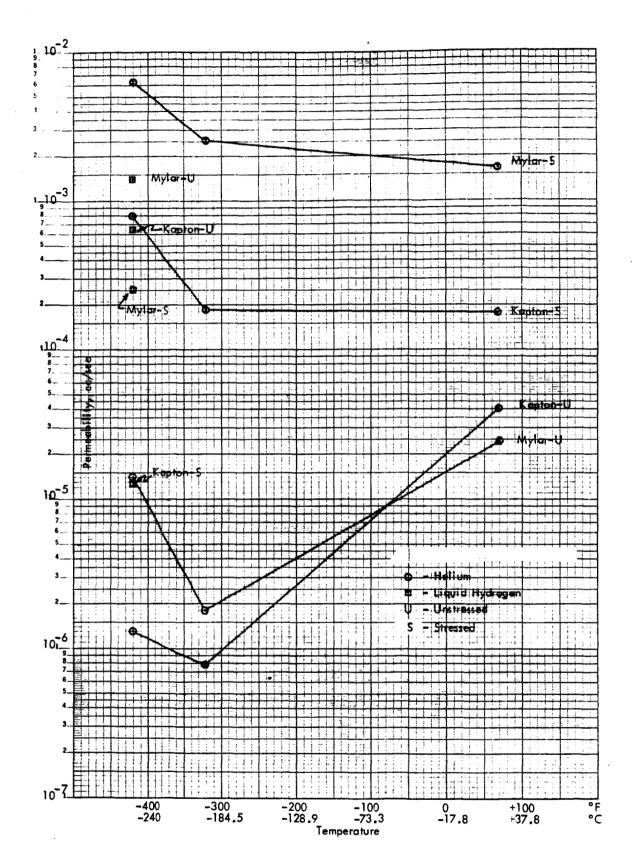


FIGURE 6- AVERAGE PERMEABILITY - 0.50 MIL (0.0127 MM) FILMS

TABLE I AVERAGE SPECIFIC PERMEABILITY RATES OF MYLAR AND KAPTON FILM

			ds .	Specific Permeability Rates	eability Ro	ıtes		
	10 <sub>3</sub> 8	Std. cc-mil/	Std. cc-mil/100 in 2.24 hrs. Atm	hrs-Atm	6-01	Std. cc-mm/cm <sup>2</sup> sec.cm Ha	cm <sup>2</sup> sec.cm	유
Maleria	+51°C	-195°C	-252°C	-252°C		-195°C	-252°C	-252°C
Fluid	GHe	GHe	GHe	<sup>Z</sup> H7	GHe	LH <sub>2</sub>	LH2	LH <sub>2</sub>
1. Mylar, .25 mil (.00635 mm) Unstressed	1.91	01	49.8	125	4.11	09	298	750
2. Mylar, .25 mil (.00635 mm) Stressed	9.13	ı	ı	ı	54.8	ı	ı	ı
3. Mylar, .5 mil (.0127 mm) Unstressed	0.0396	0.0033	0.0231	2.310	0.238	0.018	0.138	13.90
4. Mylar, .5 mil (.0127 mm) Stressed	2.8	4.29	10.2	0.413	16.8	25.7	61	2.48
5. Kapton, .25 mil (.00635 mm) Unstressed	0.622	0.349	16.5	58.0	3.73	2.09	66	324
6. Kapton, .25 mil (.00635 mm) Stressed	15.8	75	216	ဗ	95	624	1300	50
7. Kapton, .5 mil (.0127 mm) Unstressed	990.0	0.00127	0.00215	1.060	0.396	0.00762	0.0129	6.35
8. Kapton, .5 mil (.0127 mm) Stressed	0.289	0.305	1.32	0.0215	1.73	1.83	7.92	0.129
9. DuPont Data - Mylar - 1 mil	0.167	ı	ı	ı	<b>-</b>	ı	j.	ı
10. DuPont Data - Kapton - 1 mil	0.415	ı	•	ı	2.49	1	1	1

							Permeability	ty - cc/sec	
	Thickness,	Specimen	Film	Stress	ΔР		Gaseous Helium	u	LH2
Film	mils (mm)	Number	0	20%	psi (Kg/sq.cm)	+70°F(+21°C)	-320°F(-195°C)	-423°F(-252°C)	-423°F(-252°C)
Mylor	25(_00635) P1 /4M	p1/4M -1	×		1,0 (.0703)	1.1 × 10 <sup>-4</sup>	$1.67 \times 10^{-2}$	6.54 × 10 <sup>-3</sup>	<u>A</u>
		-2	×			$2.5 \times 10^{-4}$	$2.25 \times 10^{-2}$	$1.22 \times 10^{-1}$	1.46 × 10-1
	<u>-</u>	1 67	×			$2.11 \times 10^{-4}$	9.5 × 10-4	$2.96 \times 10^{-3}$	5.93 × 10 <sup>-3</sup>
		4-	×			$8.34 \times 10^{-3}$	1.79 × 10 <sup>-3</sup>	$1.35 \times 10^{-2}$	2.2
		-5	×			$4.6 \times 10^{-4}$	1.78 × 10 <sup>-2</sup>	1.54 × 10	2.94 × 10
		9-		×	-	$2.27 \times 10^{-3}$	<u>_</u>	£	<u>^</u>
		-7		×	1.0 (.0703)	$6.33 \times 10^{-2}$	<u></u>	<b></b>	<u> </u>
		8		×	0.5 (.0351)	$1.69 \times 10^{-4}$			
		6-		×	0.5 (.0351)	$1.26 \times 10^{-4}$		—	
		-10		×	0.5 (.0351)	$5.56 \times 10^{-4}$	Δ	Δ	<u> </u>
VYY Or	507 0127)	P1 /2M -1	×		1.0 (.0703)	5.1 × 10 <sup>-7</sup>	1.38 × 10-6	3.58 × 10 <sup>-6</sup>	8.2 × 10 -4
-		-2	×			$2.24 \times 10^{-6}$	4.09 × 10 <sup>-7</sup>	$1.96 \times 10^{-6}$	6.5×10 <sup>-4</sup>
		က	×			7.0 × 10 <sup>-5</sup>	2.91 × 10 <del>-</del> 6	$7.95 \times 10^{-5}$	
		4-	×	•		$1.86 \times 10^{-3}$	7.0 × 10-/	9.55 × 10 <sup>-7</sup>	$2.35 \times 10^{-4}$
		-5	×			$5.59 \times 10^{-5}$	1.85 × 10-6	1.35 × 10 <sup>-6</sup>	4.0 × 10 <sup>-3</sup>
		9-	X ×	•		$2.47 \times 10^{-6}$	1	1	1
	_	-7	×			$3.76 \times 10^{-5}$	,	1	ı
		<u>چ</u>	×		_	9.55 × 10 <sup>-6</sup>	$2.32 \times 10^{-6}$	1.04 × 10 <sup>-6</sup>	1
		6-	×		1.0 (.0703)	$2.07 \times 10^{-5}$	8.1 × 10 <sup>-7</sup>	1	
		-10	7	×	1.5 (.1055)	$3.4 \times 10^{-5}$	1.94 × 10-6	$1.52 \times 10^{-6}$	9
		-		×		$4.62 \times 10^{-3}$	$2.36 \times 10^{-3}$	$5.43 \times 10^{-3}$	3.27 × 10 <sup>-5</sup>
		-12		×	-	9.75 × 10-4	$7.52 \times 10^{-3}$	<u>^</u>	<u></u>
		-13		×		7.0 × 10-4	1.67×10 <sup>-3</sup>	1.0 × 10 <sup>-2</sup> 2	1.4 × 10-4
	<del></del>	-14		×		♠	<u>~</u>	∆ <sub>x</sub>	<u></u>
		-15		∆ ×		$1.67 \times 10^{-2}$	9.0 × 10 <sup>-3</sup>	<u>^</u>	<u>^</u>
-	<b>→</b>	-16		<u>\</u>	7 1,5 (5,1055)	2.42 × 10 <sup>-3</sup>	~01 × 12.1	1.03 × 10 <sup>-2</sup>	- 1.2 × 2.1
		, i		\ \	(200.1)				

Samples tore - unable to obtain readings under stressed conditions. A

Excessive leak in specimen.

Extra specimens.

14

TABLE II PERMEABILITY TEST MEASUREMENTS (Continued)

Thickness, Spec Film mils (mm) Num Kapton .25(.00635)P1/4K						Поп	eous Helium	
		프	Stress	<b>⊿</b>		Caseous Helium		LH2
	Number	0		psi (Kg/sq.cm)	+70° F(+21°C)	-320°F(-195°C)	-423°F(-252°C)	-423°F( -252°C)
	P1/4K -1	×		1.0 (.0703)	$2.78 \times 10^{-4}$	4.63 × 10 <sup>-4</sup>	2.78 × 10 <sup>-3</sup>	3.08 × 10 <sup>-3</sup>
	-2	×	•		1.6 × 10 <sup>-4</sup> ,	3.8 × 10-6	1	ı
	ဗ	×		<u> </u>	$1.18 \times 10^{-3}$	6.06 × 10 <sup>-5</sup>	4.66 × 10 <sup>-5</sup>	$3.4 \times 10^{-5}$
-	4-	×			$1.51 \times 10^{-3}$	8.6 × 10 <sup>-4</sup>	$7.52 \times 10^{-2}$	$2.68 \times 10^{-1}$
	-5	×		-	$6.32 \times 10^{-4}$	1.74 × 10-4	$4.65 \times 10^{-3}$	$5.45 \times 10^{-3}$
	9-		×	1.0 (.0703)	$1.19 \times 10^{-2}$	$6.7 \times 10^{-1}$	10.5	14.0
	-7		×		$5.33 \times 10^{-4}$	$1.92 \times 10^{-3}$	1.63 × 10 <sup>-2</sup>	3.31 × 10 <sup>-3</sup>
	8-		×		$2.32 \times 10^{-3}$	$1.2 \times 10^{-2}$	ī	ı
	6-		×	-	$3.5 \times 10^{-3}$	$5.22 \times 10^{-3}$	$6.32 \times 10^{-3}$	$3.38 \times 10^{-3}$
	-10		×	1.0 (.0703)	$6.8 \times 10^{-2}$	$2.2 \times 10^{-2}$	$5.6 \times 10^{-2}$	$2.3 \times 10^{-2}$
~					Ļ	•	ī	
Kapton (.50(.0127) P1/2K	P1/2K -1	×		1.0 (.0703)	$5.72 \times 10^{-5}$	$7.82 \times 10^{-7}$	3.35 × 10-7	$4.2 \times 10^{-4}$
	-2	×			$5.06 \times 10^{-3}$	8.15 × 10 <sup>-</sup> /	2.54 × 10 <sup>-0</sup>	$2.8 \times 10^{-3}$
	ငှ	×		-	$3.62 \times 10^{-3}$	$6.2 \times 10^{-7}$	1.35 × 10 <sup>-6</sup>	$1.3 \times 10^{-3}$
	4-	×			$1.065 \times 10^{-5}$	$3.6 \times 10^{-7}$	8.0 × 10-7	1.95 × 10 <sup>-5</sup>
	-5	×		1.0 (.0703)	$4.5 \times 10^{-5}$	1.29 × 10-6	1.48 × 10 <sup>-6</sup>	1.3 × 10-3
	9-		×	(\$\$01,0) \$.1	$7.3 \times 10^{-5}$	,		,
	-7		×		$3.82 \times 10^{-5}$	$2.58 \times 10^{-6}$	5.17 × 10 <sup>-6</sup>	3.6 × 10 <sup>-6</sup>
	8-		×		$6.5 \times 10^{-5}$	$9.2 \times 10^{-2}$	2.0	Tore
	6-		×	. ———	$1.57 \times 10^{-5}$	ı	ı	1
	- 10		×		$9.81 \times 10^{-3}$	$2.57 \times 10^{-6}$	>10	>10
	11-		×		$1.27 \times 10^{-3}$	1.04 × 10 <sup>-3</sup>	$3.65 \times 10^{-3}$	$5.2 \times 10^{-3}$
	-12		×		1.39 × 10 <sup>-4</sup>	1.13 × 10 <sup>-5</sup>	2.76 × 10 <sup>-6</sup>	4.0 × 10-6
-	-13		×	1.5 (.1055)	ı	1	1.0	01 🗸

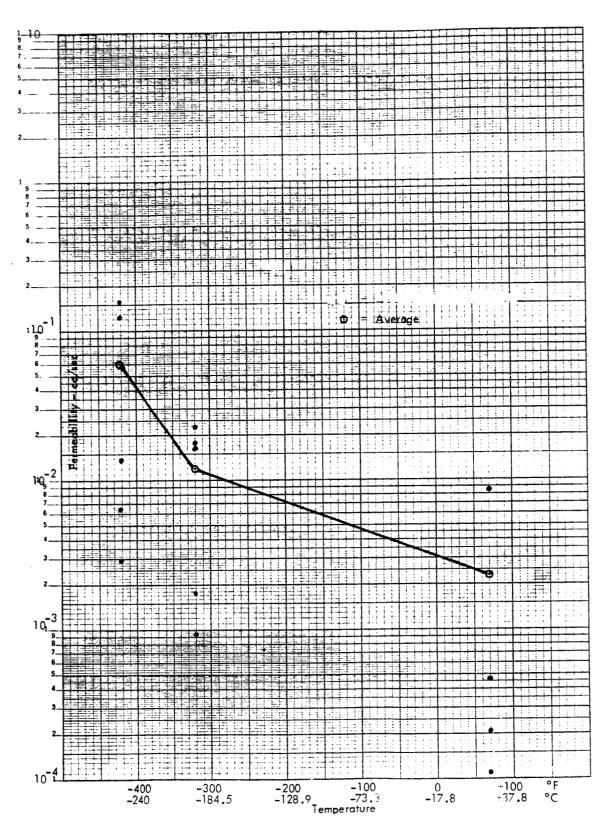


FIGURE 7 - HELIUM PERMEABILITY - 0.25 MIL (0.00635 MM) UNSTRESSED MYLAR

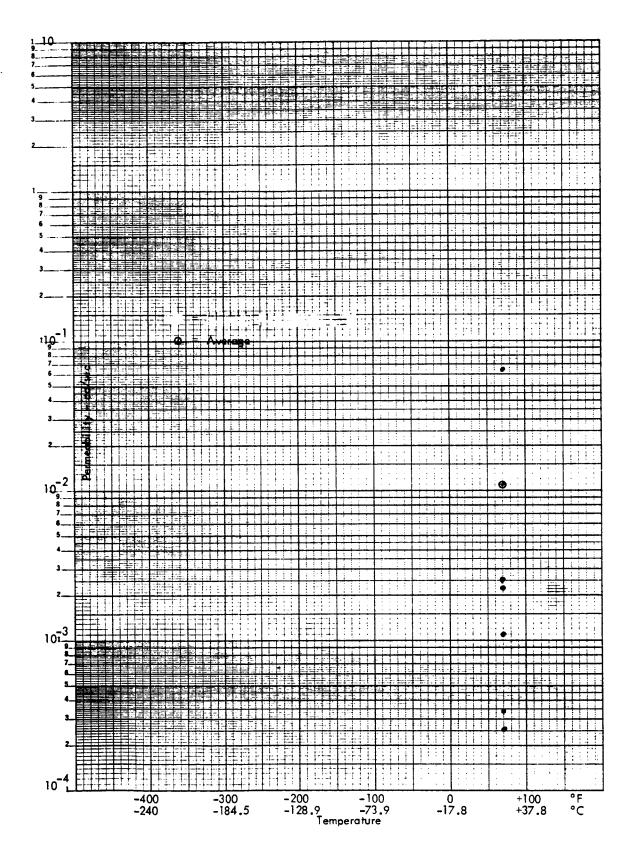


FIGURE 8 - HELIUM PERMEABILITY - 0.25-MIL (0.00635 MM) STRESSED MYLAR

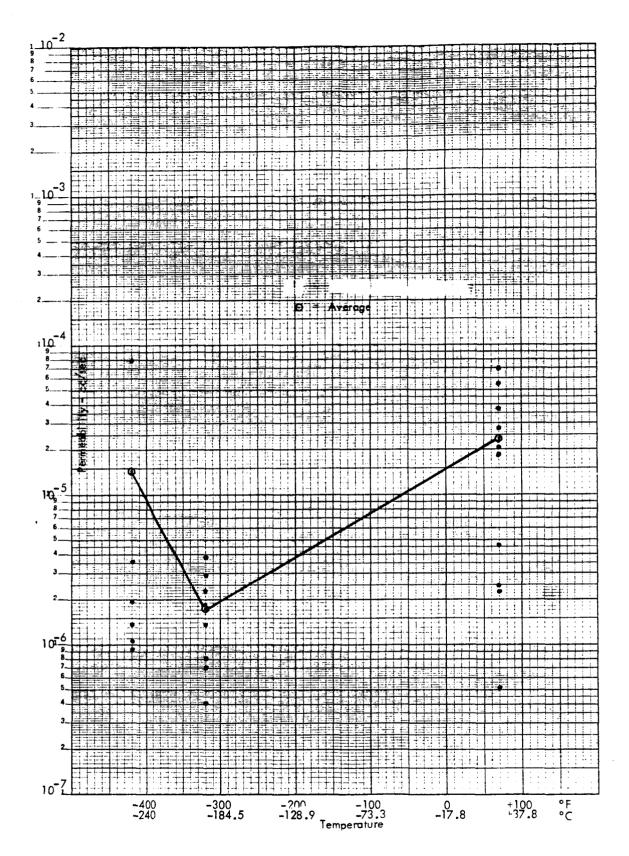


FIGURE 9 - HELIUM PERMEABILITY - 0.50-MIL (0.0127 MM) UNSTRESSED MYLAR

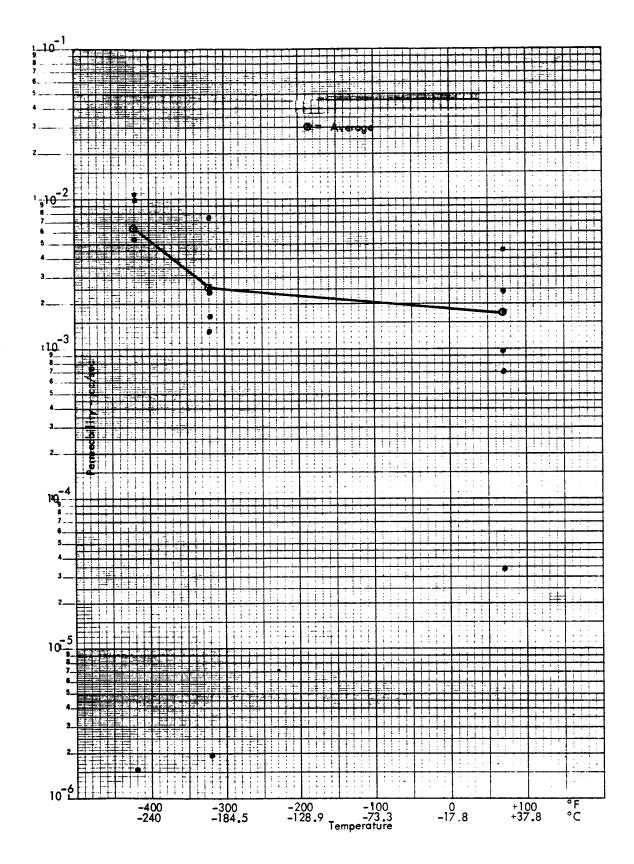


FIGURE 10 - HELIUM PERMEABILITY - 0.50-MIL (0.0127 MM) STRESSED MYLAR

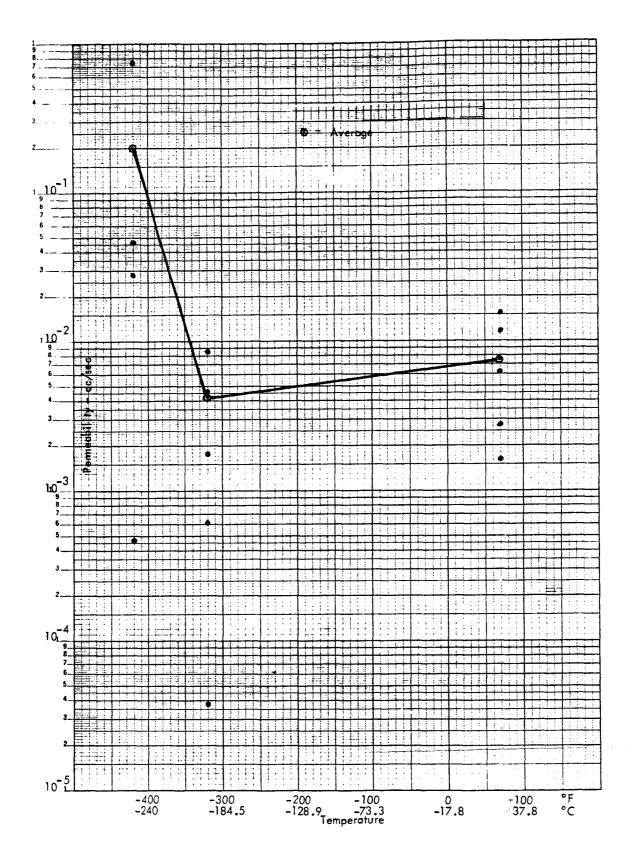


FIGURE 11 - HELIUM PERMEABILITY - 0.25-MIL (0.00635 MM) UNSTRESSED KAPTON

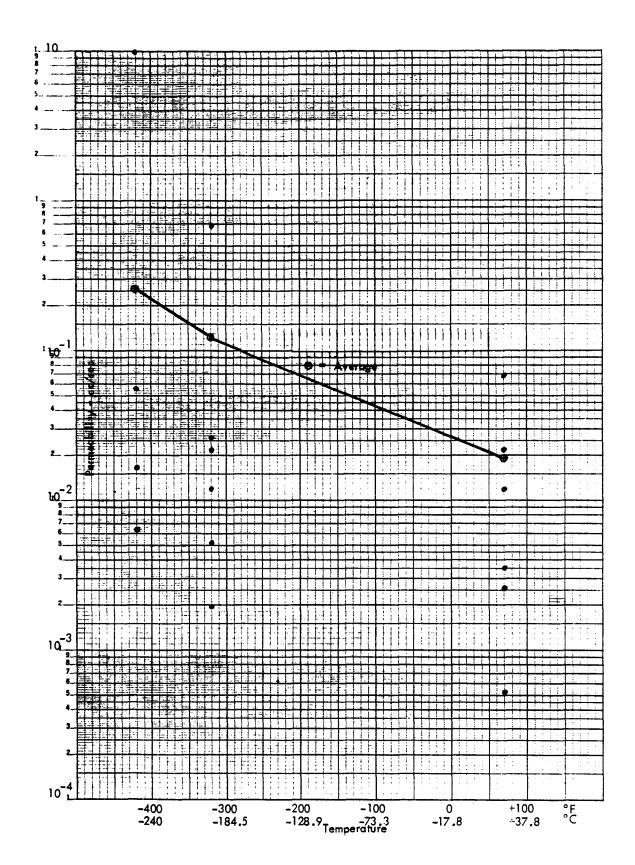


FIGURE 12 - HELIUM PERMEABILITY - 0.25-MIL (0.00635 MM) STRESSED KAPTON

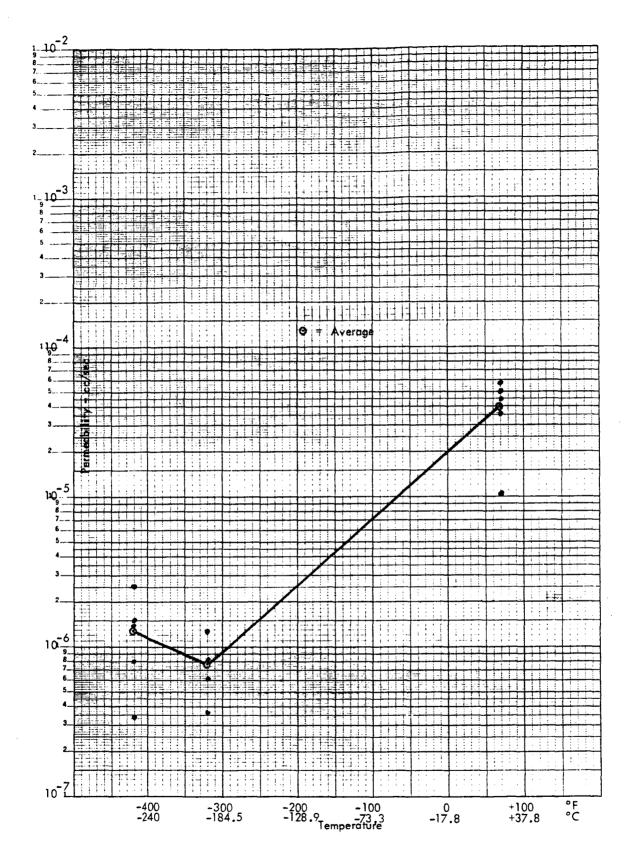


FIGURE 13 - HELIUM PERMEABILITY - 0.50-MIL (0.0127 MM) UNSTRESSED KAPTON

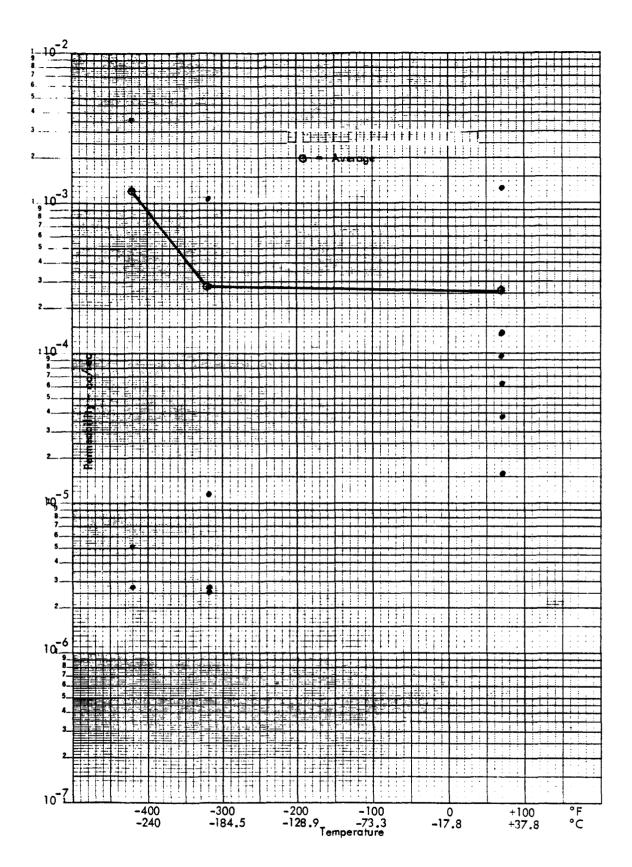


FIGURE 14 - HELIUM PERMEABILITY - 0.50-MIL (0.0127 MM) STRESSED KAPTON

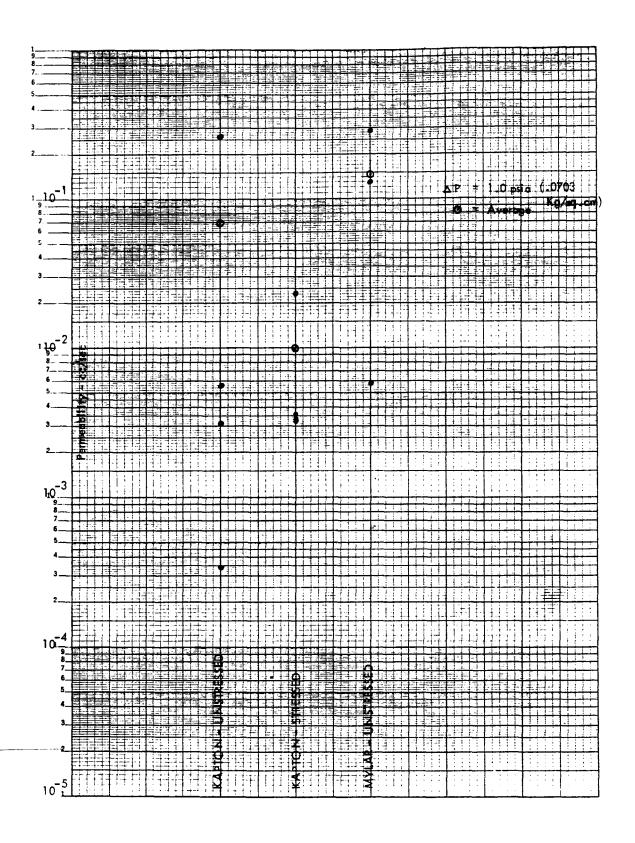


FIGURE 15 - LIQUID HYDROGEN PERMEABILITY - 0.25-MIL (0.00635 MM) FILMS

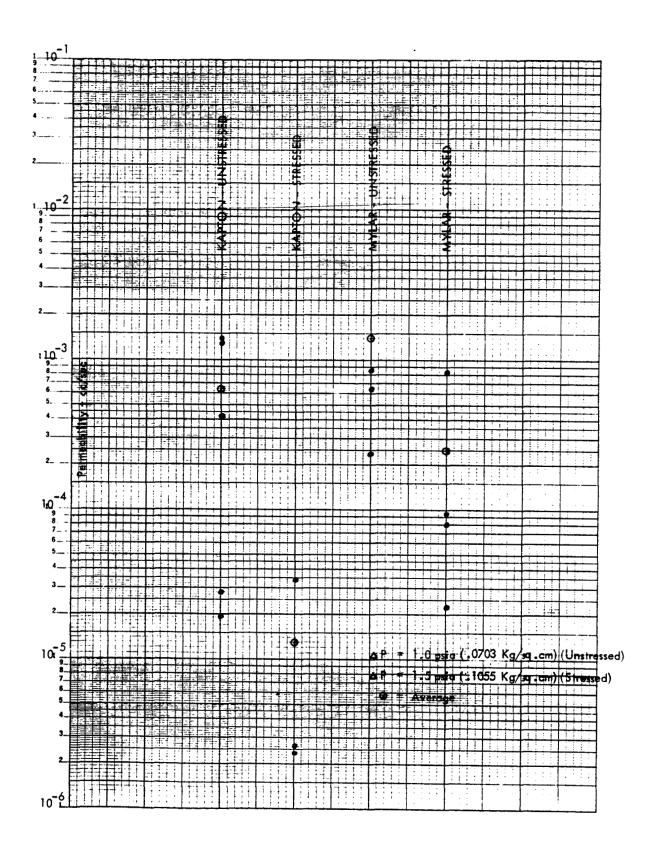


FIGURE 16 - LIQUID HYDROGEN PERMEABILITY - 0.50-MIL (0.0127 MM) FILMS

performed well with a 51.71 mm Hg (1 psi)  $\triangle$  P. No further tests were performed on the 1/4-mil Mylar.

Under all conditions the Mylar and Kapton films exhibited almost identical permeability trends. The following analyses and conclusions therefore apply to both films unless otherwise noted.

Except for the unstressed 1/4-mil Kapton in which the -195°C (-320°F) permeability was about the same as that at room temperature, the helium permeability increased with a decrease in temperature in all the 1/4-mil films. The same was true of the 1/2-mil films between -195°C and -252°C. However, the 1/2-mil unstressed films showed a decrease in helium permeability with decreasing temperature to -195°C whereas the 1/2-mil stressed film helium permeabilities were about the same at room temperature and -195°C.

Stressing the films to 20% of ultimate strength increases the helium permeability 1 to 3 orders of magnitude while at the same time decreasing the film's permeability to liquid hydrogen at -252°C.

At -252°C both film thickness evaluated are more permeable to  $LH_2$  than to  $GH_e$ . However, when stressed the inverse is true with the films being more permeable to the  $GH_e$ .

There was less data scatter and lower specific permeability in the 1/2-mil films than in the 1/4-mil films indicating better quality and uniformity. In general, the Kapton film appears to be of better quality regardless of temperature, stress, or thickness. This conclusion was also reflected in the results of the other tests (i.e., Twist-Flex, bond-strength, etc.) conducted in Task 1.

### 3.1.2 Film Flexibility

The flexibility of each film materials at cryogenic temperatures (-195°C and -252°C) was determined using the "Twist-Flex" method (References 1, 3 and 7).

The Twist-flex test apparatus is described and illustrated in detail in References 3 and 7, but briefly, the twist-flex tester rotates one of two 8.89 cm (3.5 in) diameter horizontal parallel circular plates centered on the same vertical axis. The test specimen is fastened to the plates and the system submerged in the test cryogen (liquid nitrogen or liquid hydrogen). During operation, the plates oscillate horizonally over a 90° angle while undergoing a simultaneous vertical oscillation of 3.33 cm (1.313 inches) causing a twist flexing motion as shown in Figure 17.

The twist-flex sample is nominally 10.16 cm (4 inches) wide by 27.94 cm (11 inches) long. Around the periphery on each side of the specimens a 1.25 cm (0.5 inch) strip of GT-300 heat sensitive tape was bonded to prevent crack propagations from the cut edges. Prior to testing, each sample was leak tested for defects using the procedures specified in Reference 3. Any sample showing detectable leakage ( $> 3 \times 10^{-3}$  cc/sec) was discarded.

The test specimens were each cycled for a predetermined number of cycles then removed and inspected. If a sample appeared intact then the number of cycles for the next sample was increased in an attempt to establish a threshold value. If the sample was visibly damaged then the number of cycles for the subsequent sample was reduced. After the testing was completed, the intact samples were rechecked for helium leakage and this value recorded.

The results of the Twist-Flex tests are shown in Table III. The 1/2-mil Mylar remained relatively intact up to 120 cycles at  $-195^{\circ}$ C and 15 cycles at  $-252^{\circ}$ C. The 1/4-mil Mylar went 200 cycles at  $-195^{\circ}$ C and 20 cycles at  $-252^{\circ}$ C. The Kapton films performed better than Mylar in each of the thicknesses tested. The 1/2-mil Kapton withstood 287 cycles at  $-195^{\circ}$ C and 37 cycles at  $-252^{\circ}$ C. The 1/4-mil Kapton was cycled 875 and 350 cycles at  $-195^{\circ}$ C and  $-252^{\circ}$ C respectively. The Twist-Flex test gave excellent reproducibility and even with only five samples the threshold was bracketed reasonably well. One factor that possibly helped to reduce the scatter was that no sample was Twist-Flex tested that revealed pinhole leakage (  $> 10^{-3}$  cc/sec) when checked as indicated above.

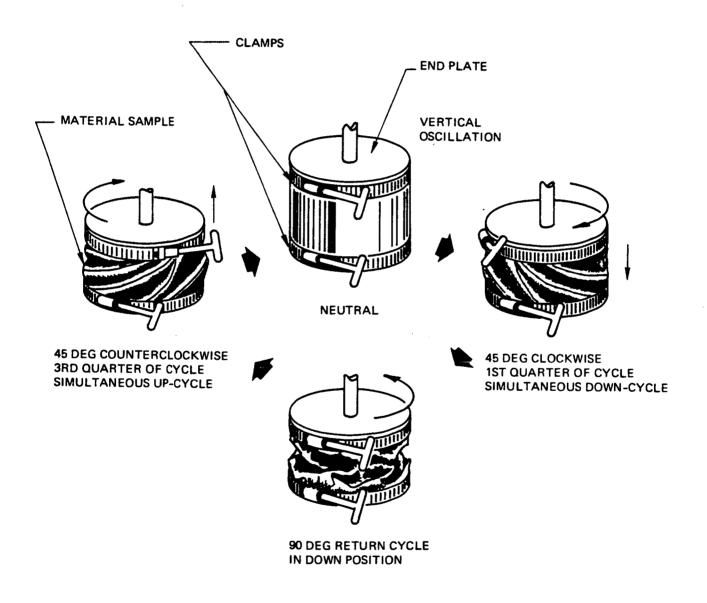


Figure 17: TWIST-FLEX ACTION

TABLE III TWIST-FLEX TESTING RESULTS

Film	Spec. No.	Film Thickness mils (mm)	Test Temp. °F(°C)	Cycles	Resul ts	Leak Rate After Test cc/sec
Mylar	T1/2M -1 -2 -3 -4 -5 -6	.5 (0.127)	-423 (-252) " " " "	100 75 50 25 15 20	Torn Torn Torn Torn Intact Torn	- - - - 17
Mylar	T1/2M -7 -8 -9 -10 -11	.5 (.0127)	-320 (-195) "	30 60 120 150 135	Intact Intact Intact Torn Torn	9.5 79 > 200
Mylar	T1/4M - 1 -2 -3 -4 -5 -6	.25 ( .00635) " " "	-423 (-252) " " "	15 25 20 22 21 20	Intact Torn Intact Torn Torn Intact	0.79 - 0 - - 0.003
Mylar	T1/4M-7 -8 -9 -10 -11	.25 ( .00635) " " "	-320 (-195) " " "	150 200 250 225 212	Intact Intact Torn Torn Torn	140 13.7 - -
Kapton	T1/2K-1 -2 -3 -4 -5 -6	.5 ( .0127)	-423 ( -252) " " " " "	20 40 25 30 35 37	Intact Torn Intact Intact Intact Intact	27.7 - 0 0.00033 No test 0.00133

TABLE III TWIST-FLEX TESTING RESULTS (Continued)

Film	Spec. No.	Film Thickness mils (mm)	Test Temp. °F (°C)	Cycles	Results	Leak Rate After Test cc/sec
Kapton	T1/2K-7 -8 -9 -10 -11	.5 (.0127)	-320 (-195)	300 250 300 275 287	Torn Intact Torn Intact Intact	- 0 - 0 0.0005
Kapton	T1/4K-1 -2 -3 -4 -5 -6	.25 ( .00635)	-423 (-252) 	90 100 200 300 423 350	Intact Intact Intact Intact Torn Intact	0.002 0.0423 0.066 0.0023 - 21.0
Kapton	T1/4K-7 -8 -9 -10 -11	.25 ( .00635)	-320 (-195) <b>V</b>	300 400 500 700 875	Intact Intact Intact Intact Intact	0.0233 0.155 0.0883 0.12 1.06

# 3.1.3 Bond Strength Tests

Bond

Bond Shear

The purpose of these tests were to establish the baseline bond strength of Mylar and Kapton film utilizing either GT-100 or GT-300 adhesives (GT Schjeldahl Co.). GT-100 is an unbacked adhesive film 0.5 mils thick and GT-300 is a Mylar (0.5 mils) backed adhesive film also 0.5 mils thick. Both are heat bondable. Butt and lap joint configurations were studied with each film material and film thickness as shown in Figure 18. A hand held sealing iron set at  $165^{\circ}\text{C} + 5^{\circ}\text{C}$  (330° + 10°F) was used to make the joints.

The results of the bond strength testing of the lap- and butt-type joints at R.T.,  $-195^{\circ}\text{C}$ ,  $-252^{\circ}\text{C}$  are tabulated in Tables IV to XI. The specimens were tested to failure in an Instron test machine using the load rate indicated in the tables.

Terminology used in describing the type of failures occurring in the bond specimens is explained below:

FAILURE TERMINOLOGY

FAILURE LOCATION	FAILURE TYPE	
Grip	Film Tensile or Tear	Film failures in the specimen grips or within 3.18 mm (0.125") of grips.
Adjacent	Film Tensile or Tear	Film failures within 3.18 mm (0.125") of joint but not within the joint itself.
Film	Film Tensile or Tear	Single film failures 3.18 mm (0.125") from joint to 3.18 mm (0.125") from grip.

All joint results were considered satisfactory in that the joints appear as strong as or stronger than the film. The results also compare favorably with the results obtained under NASA Contract NAS3-7944 (Reference 5).

Failures of the adhesive joint or film

tears within the joint.

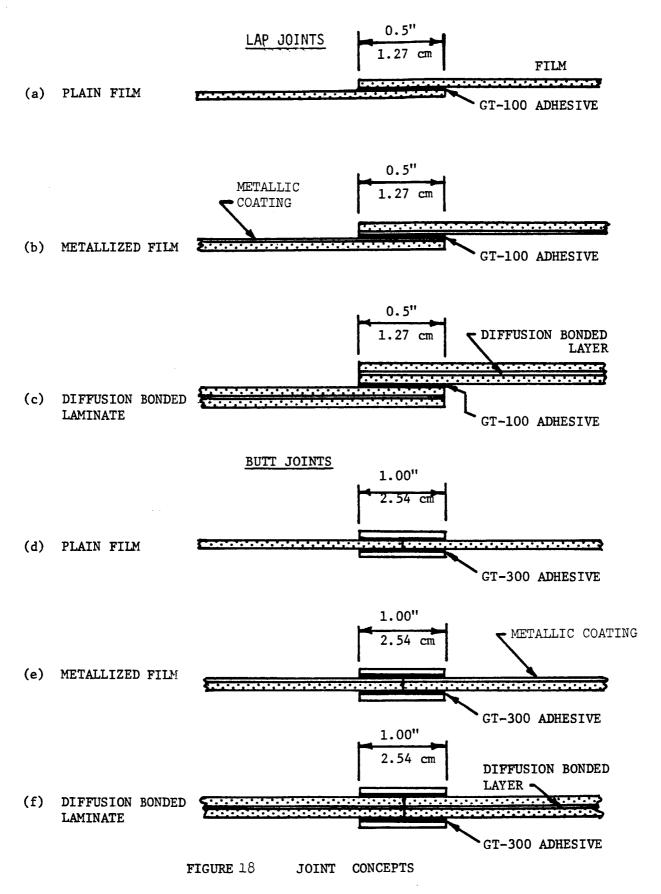


TABLE :IV BOND STRENGTH TENSILE TESTS - BUTT JOINT .25 MIL (.00635 mm) KAPTON

	Comments	<u>A</u>	<b></b>	Δ			
	Failure Location	F F F E	Film Film	Grip Adjacent Adjacent	Film	Film Adjacent Grip	Grip Grip Gr
	Joint Stress Psi (Kg/sq.cm)	4.9 (.34) 5.0 (.35) 5.6 (.39)	1.2 (.08)	7.6 (.53) 7.9 (.55) 6.1 (.43)	8.8 (.62) 7.9 (.55)	$\sim$	8.2(.5/) 7.9(.55) 7.9(.55)
	Film Stress Ksi (Kg/sq.cm)	19.6 (1378.0) 20.0 (1406.1) 22.4 (1574.9)	4.8 (337.5) 18.8 (1321.8)	30.4 (2137.3) 31.6 (2221.7) 24.4 (1715.5)	35.2 (2474.8) 31.6 (2221.7)	42.0 (2952.9) 40.0 (2812.3) 30.4 (2137.3)	32.8 (2306.1) 31.6 (2221.7) 31.6 (2221.7)
	Load Lbs (Kg)	4.9 (2.22) 5.0 (2.27) 5.6 (2.54)	1.2 (0.54)	7.6 (3.45) 7.9 (3.58) 6.12 (2.78)	8.8 (3.99) 7.91 (3.58)		8.2 (3.71) 7.9 (3.58) 7.9 (3.58)
	Test Temp., °F (°C)	R		-320 (-195)	<b></b>	-423 (-252)	
4-	Width In. (cm)	.50 (1.27)   1.00(2.54)					•
Joint	Lap (Cm)	.50 (1.27)					
	Specimen No.	B1/4K-7 -8 -9	01-	B1/4K-12 -13 -14	-15 -16	B1/4K-1 -2 -3	4 · 4

Each side (overall) lap area = 1 sq. in. (6.45 sq. cm)

Gage Lyngth = 10.0 in. (25.4 cm); Load Rate = 0.5"/min (1.27 cm/min).

 $\rightarrow$  Gage Length = 5.5 In (13.97 cm); Load Rate = 0.05"/min (.127 cm/min).

TABLE - V BOND STRENGTH TENSILE TESTS - BUTT JOINT .50 MIL (.0127 mm) KAPTON

	Joint	11						
Specimen No.	Lap (Cm)	Width In. (cm)	Test Temp., °F(°C)	Load Lbs (Kg)	Film Stress Ksi (Kg/sq.cm)	Joint Stress Psi (Kg/sq.cm)	Failure Location	Comments
B1/2K-7	.5(1.27)		RT	9.7 (4.39)	19.4 (1363.9)	(89.) 7.9	Film	<2
φ				8.3 (3.76)	16.6 (1167.1)	8.3 (.58)	Film	7 —
6-				11.4 (5.17)	22.8 (1602.9)	11.4 (.80)	Film	
-10				10.6 (4.81)	21.2 (1490.5)	10.6 (.75)	Film	
=			<b>&gt;</b>	10.1 (4.58)	20.2 (1420.2)	10.1 (.70)	Film	<b>→</b>
R1/2K_12			-320 (-195)	20.0 (9.07)	40.0 (2812.3)	20.0(1.41)	<u>E</u>	<u>^</u>
113				14.6 (6.62)	29.2 (2053.0)	14.6(1.03)	Adjacent	7
-14				12.4 (5.62)	24.8 (1743.6)	12.4 (.87)	Grip	
-15				16.7 (7.57)	33.4 (2348.3)	16.7 (1.17)	Grip	
91-			<b>→</b>	20.0 (9.07)	40.0 (2812.3)	20.0(1.41)	Bond	
B1/2K-1			-423 (-252)	22.7 (10.29)	45.4 (3192.0)	22.7 (1.59)	Bond	
-5				19.9 (9.02)	39.8 (2798.2)	19.9 (1.40)	Adjacent	
<u>ප</u>				21.4 (9.70)	42.8 (3009.1)	21.4 (1.50)	Film	
4				19.3 (8.75)	38.6 (2714.0)	19.3 (1.36)	Grip	
-5				21.1 (9.57)	42.2 (2967.0)	21.1 (1.48)	Film	
9	<b>→</b>	<b>→</b>	<b>&gt;</b>	8.6 (3.90)	17.2 (1209.3)	8.6 (.60)	Film	<b>&gt;</b>
						L		

 $\rightarrow$  Each side (overall) lap area = 1 sq. in. (6.45 sq. cm)

Gage Length = 10.0" (25.4 cm); Load Rate = 0.5"/min (1.27 cm/min)

 $\Rightarrow$  Gage Length = 5.5" (13.97 cm); Load Rate = 0.05"/min (0.127 cm/min).

TABLE VI BOND STRENGTH TENSILE TESTS - BUTT JOINT .25 MIL (.00635 mm) MYLAR

	Comments	Δ	7			<b>&gt;</b>		7								<b>→</b>
	Location	Film	Film	Film	Adjacent	Film	Grip	Grip	Grip	Grip	Grip	Grib	Grip	Film	Grip	Grip
	Psi (Kg/sq.cm)	4.0 (.28)	3.8 (.27)	4.6(.32)	3.8 (.27)	4.3 (.30)	6.1 (.43)	7.2 (.51)	8.8 (.62)	7.6 (.53)	8.6 (.60)	6.9 (.48)	11.4 (.80)	10.4 (.73)	9.4 (.66)	1.8 (.13)
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	Ksi (Kg/sq.cm)	16.0 (1124.9)	15.2 (1068.6)	18.4 (1293.6)	16.2 (1138.9)	17.2 (1209.3)	2.4 (1715.5)	28.8 (2024.8)	35.2 (2474.8)	30.4 (2137.3)	34.4 (2418.6)	27.6 (1940.5)	45.6 (3206.0)	41.6 (2924.8)	37.6 (2643.5)	7.2 (506.2)
	Lbs (Kg)	4.0 (1.81)	3.8 (1.72)	4.6 (2.09)	3.8 (1.72)	4.3 (1.95)	6,1 (2.77)	7.2 (3.27)	8.8 (3.63)	7.6 (3.45)	8.6 (3.90)	6.9 (3.13)	11.4 (5.17)	10.4 (4.72)	9.4 (4.26)	1.8 (.82)
Teet Toma	°F(°C)	RT				•	-320 (-195)				•	-423 (-252)	-	_		<b>&gt;</b>
nt Width	In. (cm)	1.00(2.54)														<b>+</b>
Joint	In. (cm)	.5(1.27)														<b>&gt;</b>
	Specimen No.	B1/4M-7	ထု	٩	-10	-	B1/4M-12	-13	-14	-15	-16	B1/4M-1	-5	ņ	4	ψ

Each side (overall) lap area = 1 sq. in (6.45 sq. cm) 

Gage Length = 10.0" (25.4 cm); Load Rate = 2.0"/min (5.08 cm/min).

Gage Length = 5.5" (13.97 cm); Load Rate = 0.05"/min (0.127 cm/min).

TABLE VII BOND STRENGTH TENSILE TESTS - BUTT JOINT .50 MIL (.0127 mm) HYLAR

Specimen No. In. (cm) In. (cm)  B1/2M-7 -9 -10 -11 B1/2M-12 -13 -14 -15 -16	Test Temp., °F (°C)  RT  RT  -320 (-195)	Load Lbs (Kg) 12.7 (5.76) 14.7 (6.67) 15.5 (7.03) 14.2 (6.44) 11.9 (5.40)	Film Stress Ksi (Kg/sq.cm) 25.4 (1785.8) 29.4 (2067.0) 31.0 (2179.5) 28.4 (1996.7) 23.8 (1673.3)	Joint Stress Psi (Kg/sq.cm) 12.7 (.89) 14.7 (1.03) 15.5 (1.09) 14.2 (1.00)	Failure Location	
.5 (1.27) 1.00(2.54)	RT		25.4 (1785.8) 29.4 (2067.0) 31.0 (2179.5) 28.4 (1996.7) 23.8 (1673.3)	12.7 (.89) 14.7 (1.03) 15.5 (1.09) 14.2 (1.00)	E.	Comments
B1/2M-12 -10 -11 -13 -14 -15 -16	-320 (-195)		29.4 (2067.0) 31.0 (2179.5) 28.4 (1996.7) 23.8 (1673.3)	14.7 (1.03) 15.5 (1.09) 14.2 (1.00)		8
-9 -10 -11 B1/2M-12 -13 -14 -15 -16	-320 (-195)		31.0 (2179.5) 28.4 (1996.7) 23.8 (1673.3)	15.5 (1.09)	Adjacent	7—
-10 -11 -13 -13 -14 -15 -16	-320 (-195)		28.4 (1996.7) 23.8 (1673.3)	14.2 (1.00)	Adjacent	
B1/2M-12 -13 -14 -15 -16	-320 (-195)		23.8 (1673.3)	•	Adjacent	
B1/2M-12 -13 -14 -15 -16	-320 (-195)	18.5 (8.16)		11.9 (.83)	Film	<b>&gt;</b>
-13 -14 -15 -16			37.0 (2601.4)	18.5(1.30)	Adjacent	
-14 -15 -16 -18		19.2 (8.71)	38.4 (2699.8)	19.2 (1.35)	Grio	7—
-15 -16 R1/2M-1		17.3 (7.84)	34.6 (2432.6)	17.3 (1.22)	Grip	
	-	22.8 (10.34)	45.6 (3206.0)	22.8 (1.60)	Adjacent	
	-	25.0 (11.34)	50.0 (3515.4)	25.0 (1.76)	Adjacent	
	-423 (-252)	18.5 (8.16)	39.0 (2742.0)	18.5 (1.30)	Grip	
	,	17.7 (8.03)	35.4 (2488.9)	17.7 (1.24)	Grip	
-3		17.7 (8.03)	35.4 (2488.9)	17.7 (1.24)	Grip	
7		23.0 (10.43)	46.0 (3234.0)	23.0 (1.62)	Grip	
		9.2 (4.17)	18.4 (1265.5)	9.2 (.65)	Grip	
<b>→</b> → 9-	<b>*</b>	24.4 (11.07)	48.8 (3431.0)	24.4 (1.72)	Adjacent	<b>-</b>

Each side (overall) lap area = 1 sq. in. (6.45 sq. cm).

Gage Length = 10.0" (25.4 cm); Load Rate = 2.0"/min (5.08 cm/min).

Gage Length = 5.5" (13.97 cm); Load Rate = 0.05"/min (0.127 cm/min).

	Comments	△	△	
	Failure Location	<u> </u>	Film Bond Adjacent Film Adjacent Grip	Grip Grip Grib Film
	Joint Stress Psi (Kg/sq.cm)	9.0(.63) 12.0(.84) 13.0(.91) 15.8(1.11) 13.2(.93)	18.4 (1.29) 21.0 (1.47) 20.4 (1.41) 10.8 (.76) 19.4 (1.36) 17.6 (1.24)	16.6 (1.17) 8.0 (.56) 21.8 (1.53) 18.0 (1.26)
	Film Stress Ksi (Kg/sq.cm)	18.0 (1265.5) 24.0 (1687.4) 26.0 (1828.0) 31.6 (2221.7) 26.4 (1856.1)	36.8 (2587.3) 42.0 (2952.9) 40.8 (2868.5) 21.6 (1518.6) 38.8 (2727.9) 35.6 (2502.9)	33.2 (2334.2) 16.0 (1124.9) 43.6 (3065.3) 36.0 (2531.0)
	Load Lbs (Kg)	4.5 (.93) 6.0 (2.72) 6.5 (2.95) 7.9 (3.58) 6.6 (2.99)	9.2 (4.17) 10.5 (4.76) 10.2 (4.63) 5.4 (2.45) 9.7 (4.40) 8.8 (3.99)	8.3 (3.76) 4.0 (1.81) 10.9 (4.94) 9.0 (4.08)
	Test Temp., °F (°C)	¥ <del></del>	-320 (-195)	<b></b>
1	Width In. (cm)	1.00(2.54)		
Joint	Lap 🕎 In. (cm)	.5 (1.27)		
	Specimen No.	L1/4K-7 -8 -9 -10 -11	L1/4K-13 -14 -15 -16 -17 L1/4K-1	

Each side (overall) lap area = 1 sq. in. (6.45 sq. cm)

• Gage Length = 10.0" (25.4 cm); Load Rate = 0.5'/min (1.27 cm/min).

 $\rightarrow$  Gage Length = 5.5"(13.97 cm); Load Rate = 0.05"/min (0.127 cm/min).

TABLE :IX BOND STRENGTH TENSILE TESTS - LAP JOINT .50 MIL (.0127 mm) KAPTON

Specimen No.		Joint						
	Lap (In. (cm)	Width In. (cm)	Test Temp., °F (°C)	Load Lbs (Kg)	Film Stress Ksi (Kg/sq.cm)	Joint Stress Psi (Kg/sq.cm)	Failure Location	Comments
L1/2K-18	.5 (1.27)	1.00(2.54)	RT	11.5 (5.22)	23.0 (1617.1)	23.0 (1.62)	Adjacent	&
-19		_		12.7 (5.76)	25.4 (1785.8)	25.4(1.79)	Film	7—
-20				8.7 (3.95)	17.4 (1223.3)	17.4(1.22)	Film	
-21				10.3 (4.67)	20.6 (1448.3)	20.6 (1.45)	Adjacent	
-22			<b>&gt;</b>	10.9 (4.94)	21.8 (1532.7)	21.8 (1.53)	Grip	<b>→</b>
21 /2/ 13				1	1			
71-V2/17			-320 (-195)	17.4 (7.89)	34.8 (2446.7)	34.8 (2.45)	Adjacent	$\triangle$
<u> </u>		-		16.6 (7.53)	33.2 (2348.3)	33.2 (2.35)	Bond	7—
-14				16.7 (7.58)	33.4 (2348.2)	33.4 (2.35)	Bond	
-15			•;•	16.9 (7.67)	33.8 (2376.4)	33.8 (2.38)	Film	
-16			<b>&gt;</b>	16.1 (7.30)	32.2 (2263.9)	32.2 (2.27)	Grip	
11/2K_1				17 1 77 200			,	
- 17 / 17			(707-) 074-	10.1 (/.30)	32.2 (2263.9)	32.2 (2.27)	Bond	
7-			-	15.4 (6.98)	30.8 (2165.5)	30.8 (2.17)	Bond	
				16.6 (7.53)	33.2 (2348.3)	33.2 (2.35)	Bond	
7	-	-		12.8 (5.81)	25.6 (1799.9)	25.6 (1.80)	Film	
5	<b>&gt;</b>	<b>&gt;</b>	<b>&gt;</b>	15.1 (6.85)	30.2 (2123.3)	30.2 (2.12)	Adjacent	-

Each side (overall) lap area = 1 sq. in. (6.45 sq. cm)

Gage Length = 10.0" (25.4 cm); Load Rate = 0.5"/min (1.27 cm/min).

• Gage Length = 5.5" (13.97 cm); Load Rate = 0.05"/min (0.127 cm/min).

BOND STRENGTH TENSILE TESTS - LAP JOINT .25 MIL (.00635 mm) MYLAR TABLE X

	Joint						
	Width	Test Temp.	Load	Film Stress	Joint Stress	Failure	
In. (cm)	In. (cm)	۲۴ (۲C)	Lbs (Kg)	Ksi (Kg/sq.cm)	Psi (Kg/sq.cm)	Location	Comments
.5(1.27)	1.00(2.54)	RT	5.4 (2.45)	21.6(1518.6)	10.8 (.76)	Grip	<u>A</u>
· 		-	4.9 (2.22)	19.6 (1378.0)	69.) 8.6	Film	
			5.6 (2.54)	22.4 (1576.9)	11.2 (.78)	Grip	
			4.4 (2.00)	17.6 (1237.4)	8.8 (.62)	Film	
		<b>→</b>	3.9 (1.77)	15.6 (1096.8)	7.8 (.55)	Film	<b>&gt;</b>
		-320 (-195)	8.8 (3.99)	35.2 (2474.8)	17.6 (1.24)	Grip	<u> </u>
			8.5 (3.86)	34.0 (2390.4)	17.0 (1.20)	Grip	<u> </u>
			8.8 (3.99)	35.2 (2474.8)	17.6 (1.24)	Grip	
			5.2 (2.36)	20.8 (1462.4)	10.4 (.73)	Adjacent	
		<b>→</b>	8.5 (3.86)	34.0 (2390.4)	17.0 (1.20)	Film	
		-423 (-252)	11.6 (5.26)	46.4 (3262.2)	23.2 (1.63)	Fila	
			3.7 (1.68)	14.8 (1040.5)	7.4 (.52)	Film	
			10.7 (4.85)	42.8 (3009.2)	21.4(1.50)	Grip	
			5.2 (2.36)	20.8 (1462.4)	10.4 (.73)	Film	
<b>&gt;</b>	<b>&gt;</b>	<b>&gt;</b>	9.2 (4.17)	36.8 (2587.3)	18.4(1.29)	Film	<b>→</b>

Each side (overall) lap area = 1 sq. in. (6.45 sq. cm). Gage Length = 10.0" (25.4 cm); Load Rate = 2.0"/min (5.08 cm/min).

Gage Length = 5.5" (13.97 cm); Load Rate = .05"/min (.127 cm/min).

BOND STRENGTH TENSILE TESTS - LAP JOINT .5 MIL (.0127 mm) MYLAR TABLE XI

Film Stress Joint Stress Failure  Ksi (Kg/sq.cm) Psi (Kg/sq.cm) Location  21.2 (1490.5) 21.2 (1.49) Film 24.6 (1729.6) 32.0 (2.25) Film 33.2 (2334.2) 33.2 (2.33) Film 31.6 (2221.7) 31.6 (2.22) Adjacent 45.2 (3177.9) 45.2 (3.12) Adjacent 29.6 (2081.1) 29.6 (2.08) Grip 47.8 (3360.7) 47.8 (3.36) Adjacent 34.4 (2629.5) 37.4 (2.63) Grip 48.8 (3431.0) 48.8 (3.43) Film 49.0 (3445.0) 49.0 (3.44) Film 40.2 (2836.3) 40.2 (2.83) Film 47.4 (3332.6) 47.4 (3.33) Adjacent		Joint	+						
.5(1,27) 1.00(2.54) RT 10.6(1.81) 21.2(1490.5) 21.2(1.49) Film 12.3(5.44) 24.6(1729.6) 24.6(1.73) Film 16.0(7.26) 32.0(2249.8) 32.0(2.25) Film 16.6(7.53) 33.2(2334.2) 33.2(2.33) Film 16.6(7.53) 33.2(2334.2) 33.2(2.23) Film 16.6(7.53) 22.6(10.25) 45.2(317.9) 45.2(3.12) Adjacent 14.8(6.71) 29.6(2081.1) 29.6(2.08) Grip 23.9(10.18) 47.8(3360.7) 47.8(3.36) Adjacent 18.7(8.48) 34.4(2629.5) 37.4(2.63) Grip 24.4(11.07) 48.8(3431.0) 48.8(3.43) Film 24.4(11.07) 48.8(3431.0) 49.0(3.44) Film 24.4(11.07) 48.8(3431.0) 49.0(3.44) Film 20.1(9.12) 47.4(3.33) Adjacent 47.4(3.33) Adjacent 23.7(10.75) 47.4(3.33) Adjacent 47.4(3.33) Adjacent 23.7(10.75) 47.4(3.33) Adjacent 23.7(10.75) 47.4(3.33) Adjacent 23.7(10.75) 47.4(3.33) Adjacent 23.7(10.75) 47.4(3.33) Adjacent 47.4(3.33)	Z		Width In. (cm)	Test Temp. °F (°C)	Load Lbs (Kg)	Film Stress Ksi (Kg/sq.cm)	Joint Stress Psi (Kg/sq.cm)	Failure Location	Comments
12.3 (5.44) 24.6 (1729.6) 24.6 (1.73) F 16.0 (7.26) 32.0 (2249.8) 32.0 (2.25) F 16.6 (7.53) 33.2 (2334.2) 33.2 (2.33) F 16.6 (7.17) 31.6 (2221.7) 31.6 (2.22) 15.8 (7.17) 31.6 (2221.7) 31.6 (2.22) 14.8 (6.71) 29.6 (2081.1) 29.6 (2.08) 23.9 (10.18) 47.8 (3360.7) 47.8 (3.36) 18.7 (8.48) 34.4 (2629.5) 37.4 (2.63) 24.4 (11.07) 48.8 (3431.0) 48.8 (3.43) 24.5 (11.11) 49.0 (3445.0) 49.0 (3.44) 124.4 (11.07) 48.8 (3431.0) 48.8 (3.43) 20.1 (9.12) 47.4 (3332.6) 47.4 (3.33)	11/011	F (1 27)	1 00/2 54)	RT	10.6(1.81)	21.2 (1490.5)	21.2 (1.49)	Film	$\triangle$
16.0 (7.26)   32.0 (2249.8)   32.0 (2.25)   F     16.6 (7.53)   33.2 (2334.2)   33.2 (2.33)   F     16.6 (7.53)   33.2 (2334.2)   33.2 (2.33)   F     16.8 (7.17)   31.6 (2221.7)   31.6 (2.22)     16.8 (7.17)   31.6 (2221.7)   31.6 (2.22)     16.8 (7.17)   29.6 (2081.1)   29.6 (2.08)   47.8 (3360.7)   47.8 (3360.7)   47.8 (3.36)   47.8 (3.36)     18.7 (8.48)   34.4 (2629.5)   37.4 (2.63)     18.7 (8.48)   34.4 (2629.5)   37.4 (2.63)     24.4 (11.07)   48.8 (3431.0)   48.8 (3.43)     24.4 (11.07)   48.8 (3431.0)   48.8 (3.43)     24.4 (11.07)   48.8 (3431.0)   49.0 (3.44)     24.4 (11.07)   48.8 (3431.0)   49.2 (2.83)     20.1 (9.12)   47.4 (3332.6)   47.4 (3.33)	L1/2M =/	(/7:1) C:	(+2.2)00.1	<u> </u>	12,3 (5,44)	24.6 (1729.6)	24.6 (1.73)	Film	
16.6 (7.53)   33.2 (2334.2)   33.2 (2.33)   F     15.8 (7.17)   31.6 (2221.7)   31.6 (2.22)     -320 (-195)   22.6 (10.25)   45.2 (3177.9)   45.2 (3.12)     14.8 (6.71)   29.6 (2081.1)   29.6 (2.08)     23.9 (10.18)   47.8 (3360.7)   47.8 (3.36)     18.7 (8.48)   34.4 (2629.5)   37.4 (2.63)     18.7 (8.48)   34.4 (2629.5)   37.4 (2.63)     24.4 (11.07)   48.8 (3431.0)   49.0 (3.44)     24.5 (11.11)   49.0 (3445.0)   49.0 (3.44)     24.4 (11.07)   48.8 (3431.0)   48.8 (3.33)     20.1 (9.12)   40.2 (2836.3)   47.4 (3.33)     47.4 (3.33)   47.4 (3.33)	ρ ο		_		16.0 (7.26)	32.0 (2249.8)	32.0 (2.25)	Film	
15.8 (7.17)   31.6 (2221.7)   31.6 (2.22)   45.2 (3177.9)   45.2 (3.12)   45.2 (3177.9)   45.2 (3.12)   47.8 (6.71)   29.6 (208)   47.8 (3.36)   47.8 (3.36)   47.8 (3.36)   47.8 (3.36)   47.8 (3.43)   47.8 (3.43)   48.8 (3.43)   49.0 (3445.0)   48.8 (3.43)   49.0 (3445.0)   48.8 (3.43)   49.0 (3.44)   49.0 (3.44)   49.0 (3.44)   49.0 (3.44)   49.0 (3.44)   49.0 (3.44)   49.0 (3.43)   40.2 (2.83)   20.1 (9.12)   47.4 (3.33)   47.4	<u> </u>				16.6 (7.53)	33.2 (2334.2)	33.2 (2.33)	Film	
-320 (-195) 22.6 (10.25) 45.2 (3177.9) 45.2 (3.12) 29.6 (2.08) 23.9 (10.18) 47.8 (3360.7) 47.8 (3.36) 47.8 (3.36) 47.8 (3.36) 48.8 (3.43) 48.8 (3.43) 48.8 (3.43) 49.0 (3.44) 49.0 (3.445.0) 48.8 (3.43) 49.0 (3.445.0) 48.8 (3.43) 49.2 (2.83) 40.2 (2.83) 47.4 (3.33)				<b>→</b>	15.8 (7.17)	31.6 (2221.7)	31.6 (2.22)	Adjacent	<b>&gt;</b>
14.8 (6.71)       29.6 (2081.1)       29.6 (2081.1)       29.6 (2081.1)       29.6 (2081.1)       47.8 (3.36)       47.8 (3.36)       47.8 (3.36)       47.8 (3.36)       47.8 (3.36)       47.8 (3.36)       47.8 (3.36)       47.8 (3.36)       47.8 (3.36)       47.8 (3.36)       48.8 (3.43)       48.8 (3.43)       48.8 (3.43)       48.8 (3.44)       11.07       48.8 (3.43)       49.0 (3.44)       18.8 (3.43)       40.2 (2.83)       47.4 (3.33)         20.1 (9.12)       47.4 (3.332.6)       47.4 (3.33)       47.4 (3.33)       47.4 (3.33)	61 770/11		<del></del>	-320 (-195)	22.6 (10.25)	45.2 (3177.9)	45.2 (3.12)	Adjacent	
23.9 (10.18) 47.8 (3360.7) 47.8 (3.36) 18.7 (8.48) 34.4 (2629.5) 37.4 (2.63) 67.4 (11.07) 48.8 (3431.0) 48.8 (3.43) 724.4 (11.07) 49.0 (3445.0) 49.0 (3.44) 724.4 (11.07) 49.8 (3431.0) 48.8 (3.43) 720.1 (9.12) 40.2 (2836.3) 40.2 (2.83) 47.4 (3.33)	L1/2M -12			(2) (2)	14.8 (6.71)	29.6 (2081.1)	29.6 (2.08)	Grip	
18.7 (8.48) 34.4 (2629.5) 37.4 (2.63) 6 24.4 (11.07) 48.8 (3431.0) 48.8 (3.43) 7.4 (2.63) 6 -423 (-252) 12.7 (5.76) 25.4 (1785.7) 25.4 (1.78) 49.0 (3.44) 49.0 (3.445.0) 49.0 (3.44) 10.2 (2.83) 40.2 (2.83) 40.2 (2.83) 40.2 (2.83) 47.4 (3.33)	2 -				23.9 (10.18)	47.8 (3360.7)	47.8 (3.36)	Adjacent	
-423 (-252) 12,7 (5,76) 25.4 (1785.7) 25.4 (1.78) 49.0 (3445.0) 48.8 (3.43) 49.0 (3445.0) 49.0 (3431.0) 48.8 (3.43) 20.1 (9.12) 40.2 (2836.3) 40.2 (2.83) 47.4 (3.33)	4 7				18.7 (8.48)	34.4 (2629.5)	37.4 (2.63)	Grip	
-423 (-252) 12,7 (5,76) 25.4 (1785.7) 25.4 (1.78) 49.0 (3445.0) 49.0 (3.44) 49.0 (3445.0) 48.8 (3.43) 40.2 (2836.3) 40.2 (2.83) 47.4 (3.33)	61- 91-			<b>→</b>	24.4 (11.07)	48.8 (3431.0)	48.8 (3.43)	Adjacent	
24.5 (11.11) 49.0 (3445.0) 49.0 (3.44) 48.8 (3.43) 48.8 (3.43) 40.2 (2836.3) 40.2 (2.83) 47.4 (3.33)				-423 (-252)	12.7 (5.76)		25.4 (1.78)	Film	
24.4 (11.07) 48.8 (3431.0) 48.8 (3.43) 20.1 (9.12) 40.2 (2836.3) 40.2 (2.83) 47.4 (3.33)	LI/ZM -1			(202 ) 621	24.5 (11.11)		49.0 (3.44)	Film	
20.1 (9.12) 40.2 (2836.3) 40.2 (2.83) 47.4 (3.33)	7-				24.4 (11.07)		48.8 (3.43)	Film	
47.4 (3332.6) 47.4 (3.33)	? `				20.1 (9.12)		40.2 (2.83)	Film	
	4 4	<b>→</b>	<b>→</b>	<b>→</b>	23.7 (10.75)		47.4 (3.33)	Adjacent	<b>&gt;</b>

Each side (overall) lap area = 1 sq. in. (6.45 sq. cm). Gage Length = 10.0" (25.4 cm); Load Rate = 2.0"/min (5.08 cm/min). Gage Length = 5.5" (13.97 cm); Load Rate = 0.05"/min (0.127 cm/min).

# 3.1.4 Interply Inflation Tests

The purpose of the interply inflation tests was to assess the quantity of gas and/or cryogen diffusing through the polymeric films at cryogenic temperatures. Tests under NASA Contract NAS3-6288 (Reference 3) indicated that in times as short as 5 minutes sufficient gas permeation occurred through the polymeric films at low temperatures to cause interply inflation of bladders during warm-up.

The test samples consisted of two plies of Mylar or Kapton, 10.16 cm x 27.9 cm (4" x 11") in size, bonded together around the edges with GT-100 and -300 adhesive as shown in Figure 19 to form an enclosed envelope. Each ply of film was inspected for leakage prior to assembling the specimen to insure integrity of the film. Films showing detectable leakage (> 10<sup>4</sup> cc/sec) were discarded. The thickness of the sample was then measured (in case air was trapped in the specimen during sealing) at room temperature. To measure that thickness the entrapped gas was forced to one end of the specimen and an average thickness reading obtained. However, as the data in Tables XII and XIII shows none of the samples contained initially entrapped air. Next the sample was submerged in either liquid nitrogen or liquid hydrogen, depending on the test temperature desired, for 5 minutes. After the time period was over, the sample was immediately withdrawn from the cryogen and permitted to warm up to room temperature. If any fluid (gas or liquid) permeated the film, the sample would expand as the temperature increased. As the data in Tables XII and XIII shows, no evidence of interply inflation was noted.

Since the results of the -195°C and -252°C interply inflation tests were contrary to results previously published for the Mylar and Kapton film (References 3 and 7), several additional tests were conducted to insure that the results were valid.

First the submersion time on selected samples was increased to 30 minutes in both liquid nitrogen and liquid hydrogen and again no inflation occurred. One Mylar and one Kapton specimen was then submerged for 24 hours in liquid nitrogen with no evidence of inflation. Finally, specimens of 1/4 mil Mylar

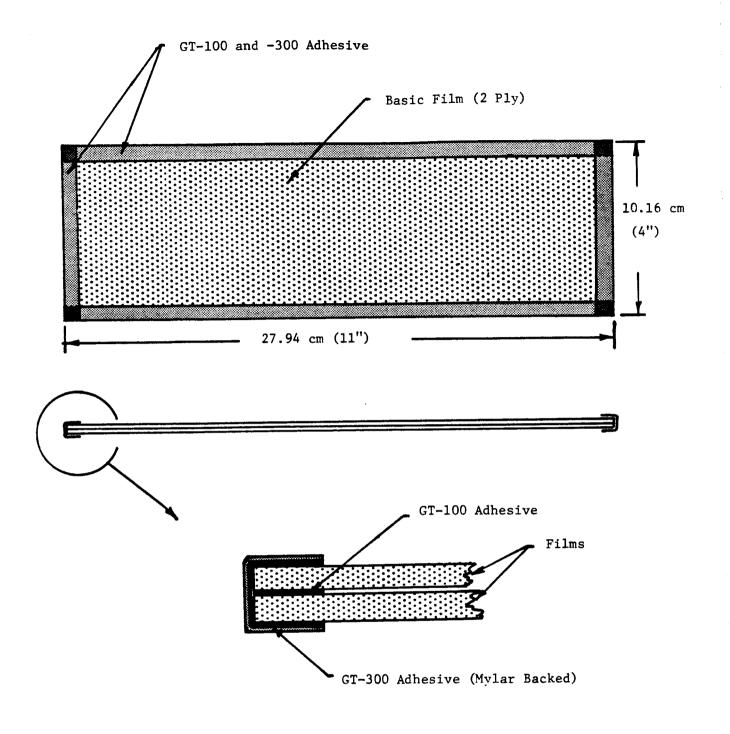


FIGURE 19 INTERPLY INFLATION SPECIMEN

NC = No Change

Thickness Inflation 24 Hrs. S Z  $\overset{\circ}{\mathsf{Z}}$ Exposure Time **Thickness** Inflation 30 Min. 991199119911 TABLE XII INTERPLY INFLATION TESTS AT -320°F (-195°C) **Thickness** Inflation 5 Min. Z. Thickness Initial Coating None Film Thickness .25 (.00635) .50 (.0127) .25 (.00635) 25 (.00635) .00635) .25 (.00635) .50 (.0127) .50 (.0127) mils (mm) Material Mylar Kapton Kapton Mylar Specimen No. 

TABLE XIII INTERPLY INFLATION TESTS AT -423°F (-252°C)

															 						<del></del>			
e Time	30 Min.	Inflation	Thickness	Z	} ◀	4		<del></del>					-		 _							<b>-</b>		O Z
Exposure Time	5 Min.	Inflation	Thickness	C Z	<u>}</u> <														<del></del>				<b>&gt;</b>	O Z
			Thickness	C	> <	<b>—</b>																	<u> </u>	0
			Coating	Q Q	υ - Σ	←								- "-									<b>→</b>	None
		Film Thickness	mils (mm)	25 ( 00635)	(55000.) 53.	<b>4</b>		<b>~</b>	.25 (.00635)	.50 (.0127)	<b>.</b>		-	.50 (.0127)	.25 (,00635)	<b>4</b>		<b>&gt;</b>	,25 (,00635)	.50 (.0127)	<b></b>		<b>&gt;</b>	.50 (.0127)
			Material		wy.iar	<b>-</b>							<b>~</b>	Mylar	Kapton	4							<b>→</b>	Kapton
			Specimen No.	2]	17	7.7	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40

NC = No Change

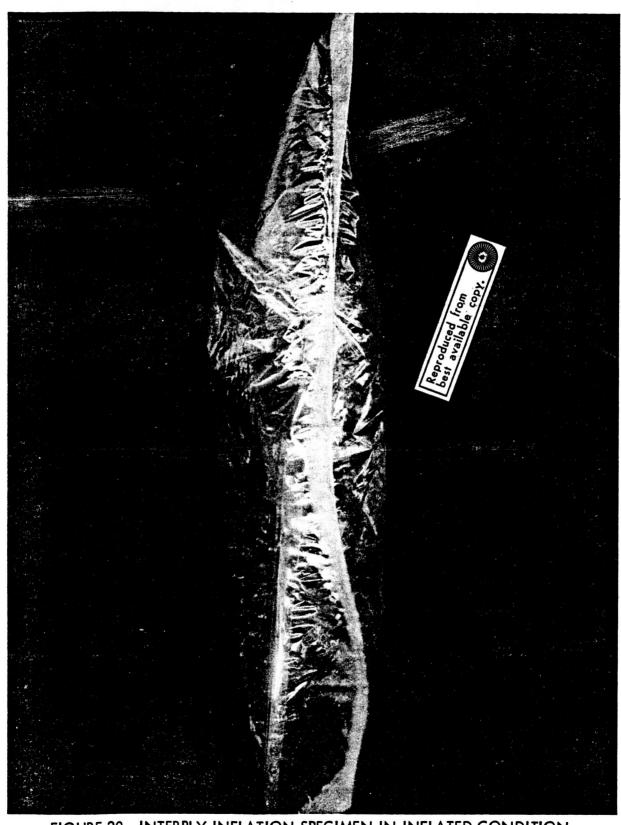


FIGURE 20 INTERPLY INFLATION SPECIMEN IN INFLATED CONDITION

and 1/4 mil Kapton were intentionally punctured with one very small pinhole and then submerged in liquid nitrogen for a period of 5 minutes. When the specimens were removed they immediately inflated to approximately 5.1 cm (2 inches) in depth, indicating inflation will occur rapidly if fluid is allowed to penetrate the film. The inflated specimen is shown in Figure 20.

From these tests it was concluded that careful screening of the film samples can minimize and possibly eliminate interply inflation. However, the slightest imperfection (pinhole, poor seam, thin spot, etc.) will permit inflation to occur. However, on a bladder of any substantial size it may be very difficult to eliminate all of these defects. It should be noted that several areas of the as-received films (both Mylar and Kapton in both thickness) had to be rejected due to pinholes and imperfections. If samples had been taken from these areas then interply inflation would have occurred in these tests.

# 3.2 Preliminary Coating Evaluation

The objective of this phase of the program was to select ten basic coating concepts and determine their effectiveness in reducing the permeability of either Mylar, Kapton or both. This work on preliminary coating evaluated constituted Task II of the program.

# 3.2.1 Coating Concepts

The coating concepts studied in this phase of the program are shown in Table XIV. Six of the concepts involved a laminate construction in which two layers of metallized Kapton film were diffusion bonded together. The remaining four concepts were metallized films (one side only).

The selection of a coating concept was based on achieving the maximum reduction in the permeability with the minimum effect on the ability of Mylar and Kapton films to perform satisfactorily in cryogenic bladder applications. The coating properties that had to be considered were: ductility, corrosion resistance, adhesion, fabricability and thickness.

It was desired that the coating be ductile and not work harden in order to provide maximum flex life. Corrosion resistance was desirable to preserve coating continuity during bladder fabrication and storage. The following metals potentially met these requirements; aluminum, copper, gold, and silver.

Good adhesion was necessary in order to maintain coating integrity, but adhesion is related to the application method as well as the coating metal and substrate.

The laminate concept (sandwich construction) indicated in Table XIV is illustrated in Fig. 21. With this concept, the Kapton film is metallized on one side by either sputtering or vacuum-deposition (other processes could be used), taking care to keep all surfaces clean. Two sheets of the metallized film are then stacked with the two metallized surfaces in contact with one another. Heat and pressure  $[300 \, ^{\circ}\text{C} \, (575 \, ^{\circ}\text{F})]$  and  $84.4 \, \text{Kg/cm}^2 \, (1200 \, \text{psi})$  for 5 minutes are then

# TABLE XIV PROPOSED COATINGS

Film Thickness, mils	1,	/4	1/2	
Coating Thickness, Å	5,000	10,000	5,000	10,000
KAPTON (for sandwich construction)				
Vacuum Deposited Gold		×		×
Sputtered Gold	Х	×	×	×
KAPTON (metallized one side)				
Sputtered Aluminum	x			×
MYLAR (metallized one side)				·
Sputtered Aluminum	×			Х

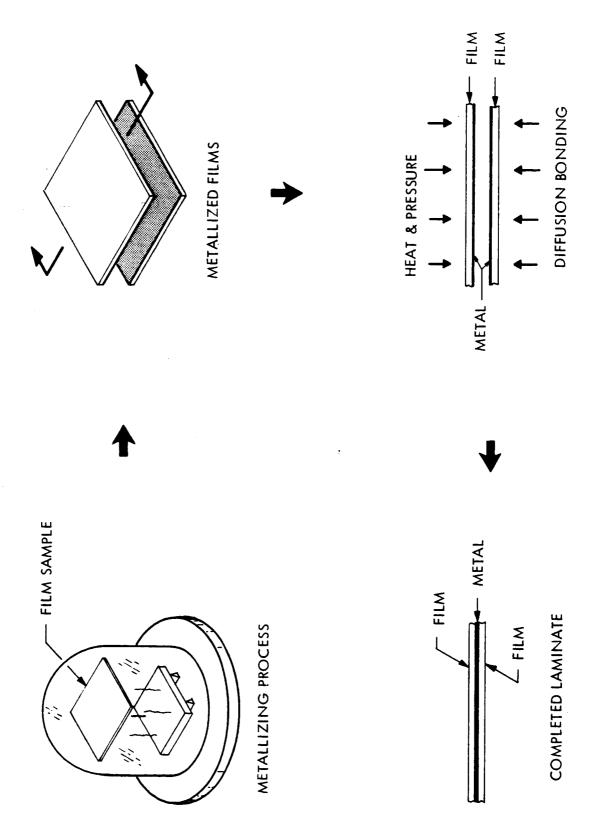


FIGURE 21 DIFFUSION BONDED LAMINATE PROCESS

applied to diffusion bond the two surfaces together. If the metallized surface is of high quality (low porosity), then the two surfaces tend to seal the pores in the other, thereby reducing the overall permeability of the film substantially. The principle is illustrated in Figure 22 with two plies of poorly metallized Kapton film. The individual metallized films have visible pores in the coating which are no longer visible in the diffusion bonded portion. Figure 23 shows the diffusion bonded interface of two layers of gold metallized Kapton.

Of the potentially useful metals for coating, gold best fulfilled the requirements of the sandwich concept. Gold is readily diffusion bonded at temperatures (and pressures) that can be withstood by the Kapton film. It is probable that processes could be developed for using aluminum, copper, or silver; however, techniques for preventing oxidation of the coatings prior to diffusion bonding would complicate processing. Therefore gold was used to evaluate this concept.

Application methods considered for applying the gold coating included sputtering, vacuum deposition, electroless plating, chemical vapor deposition, and ion beam. Sputtering and vacuum deposition were selected for this program. Sputtering has been shown to be capable of applying gold in a dense, uniform, adherent coating of controlled thickness. The method is generally sufficiently energetic to effect good adhesion between the relatively inert gold and Kapton, but for reliability a flash (500 Å) coating of chromium was initially applied to the Kapton, increasing the adherence of the metallization.

During sputtering, atoms of the coating material bombard the substrate at an extremely high velocity. Sputtered atoms have a very high energy (1-10 electron volts) as compared with other deposition techniques (typically 0.1 electron volt) so that a strong bond is made with the substrate and a dense, uniform, nonporous coating is formed. A sputtered coating is shown in Figure 24.

Vacuum deposition is a well established process resulting in uniform coatings. However, it is desirable to prepare the Kapton surface by an initial minimum deposition of an active metal, such as chromium, prior to depositing gold with the vacuum process in order to effect good adhesion. Since vacuum deposition

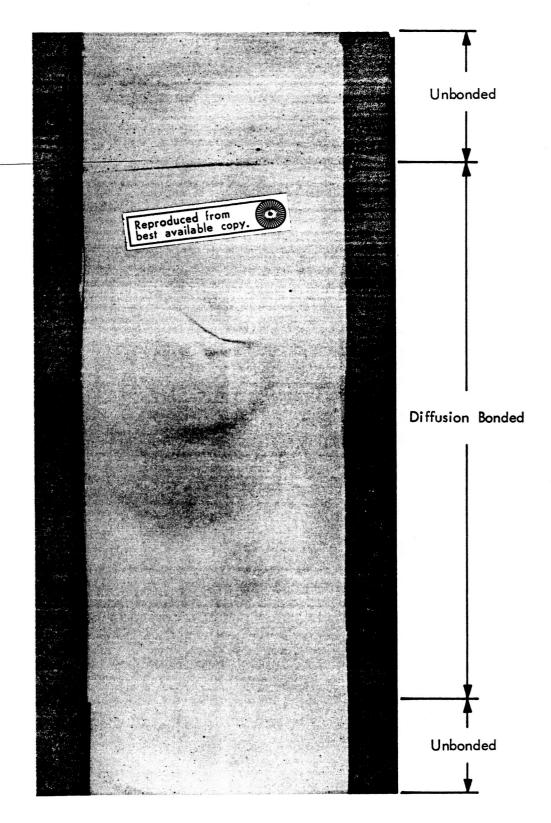


FIGURE 22 VACUUM DEPOSITED GOLD ON KAPTON FILM — AFTER DIFFUSION BONDING

# KAPTON/GOLD LAMINATE

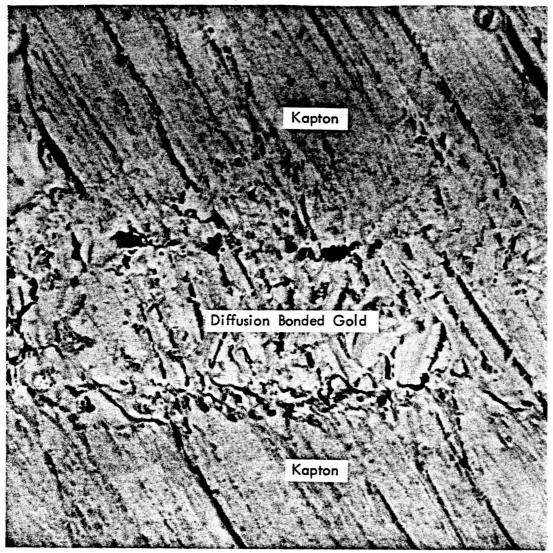


FIGURE 23 DIFFUSION BONDED GOLD-KAPTON FILM

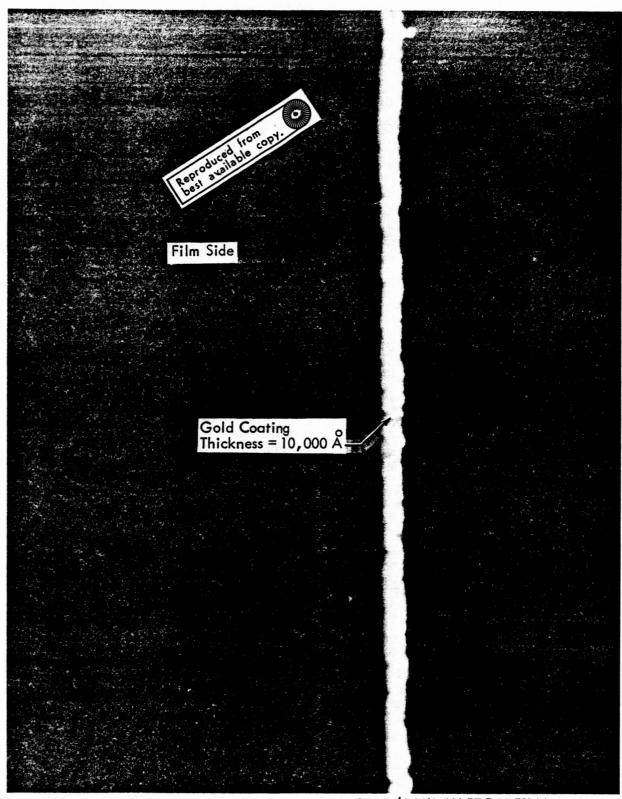


FIGURE 24 SPUTTERED GOLD COATING ON 1/4 MIL KAPTON FILM — CROSS-SECTION VIEW (5000 X)

is a reliable coating process and widely used, it was desirable to compare it with a sputtered coating with the diffusion bonding concept as shown in Table XIV.

The "sandwich" concept is not applicable to Mylar because of the relatively high temperature required for diffusion bonding, and therefore the single ply concept as indicated in Table XIV was evaluated. Sputtering provides a very thin but dense coating that is not obtainable by the other plating techniques. Aluminum was selected since it is an active metal that adheres relatively well to both Kapton and Mylar. Sputtering of aluminum on Mylar is an established process. For comparison, aluminum was also sputtered on Kapton. Both coatings had a flash of chromium (500 Å) applied to increase adherence . It appeared desirable to evaluate the performance of a single coating of aluminum on Mylar and Kapton since it represented the simplest practical approach using an available coating technique to meet the objectives of this program.

Selection of coating thickness was a compromise between a thin, flexible coating that may have, inherently, a relatively high permeability and a thick coating that would inhibit the cryogenic flexibility of the polymeric film. Based on handling, visual observations, and limited test data on coated films, a thickness of 10,000 Å appeared reasonable as a starting point for evaluating concept feasibility. This amount of metal deposition was acceptable from both the application and diffusion bonding standpoint for the sandwich concept, and has a significant thickness for reducing permeability, and does not affect the handling of the polymer film. In order to evaluate the effect of coating thickness on permeability and film performance, a second thickness of 5,000 Å was arbitrarily selected.

#### 3.2.2 Permeability Testing

The permeability of the metallized film concepts to helium gas at  $+25^{\circ}$ C was determined with the film in a state of zero and 20% stressed condition. The tests were made using the same apparatus and procedures used in the preliminary film characterization studies (Paragraph 3.1.1).

The results of the permeability tests are presented in Tables XV and XVI. Table XV shows the individual test values of the stressed and unstressed films while Table XVI presents the average specific permeability rates.

TABLE XV SPECIFIC PERMEABILITY RATES OF PLAIN AND METALLIZED FILMS

		-					Total		
	Film			Film	Coating	61	Laminate	Helium (1)	
	Stress	SS	•	Thickness		Thickness,	Thi ckness,	Leak Rate	Specific Permeability
Concept	0% 20%	%0?	Material	mils (mm)	Material	4	mils (mm)	cc/sec	cc-mm/cm-sec.cm ng
Plain Film (3)	×		Mylar	.25 (.00635)	ı	ı	.25 (.00635)	$2.3 \times 10^{-3}$	$1.14 \times 10^{-8}$
	×		Mylar	.50 (.0127)	ı	1	.50 (.0127)	2.4 × 10 <sup>-5</sup>	$2.38 \times 10^{-10}$
	×			.25 (.00635)	1	ı	.25 (.00635)	$7.5 \times 10^{-3}$	$3.73 \times 10^{-9}$
	×		Kapton	.50 (.0127)	ı	1	.50 (.0127)	4.0 × 10 <sup>-5</sup>	$3.96 \times 10^{-10}$
	×			1.00 (.0254)	ı	1	1.00 (.0254)	ı	$2.49 \times 10^{-9}$
		×	Mylar	.25 (.00635)	1	ı	.25 (.00635)	$1.1 \times 10^{-2}$	$5.48 \times 10^{-8}$
		×	Mylar	.50 (.0127)	ı	ı	.50 (.0127)	1.70 × 10 <sup>-3</sup>	$1.68 \times 10^{-8}$
£1		×	Kapton	.25 (.00635)	ı	ı	.25 (.00635)	1.9 × 10-2	9.5 × 10 <sup>-8</sup>
_			Kapton	.50 (.0127)	ı	ı	.50 (.0127)	1.75 × 10 <sup>-4</sup>	$1.73 \times 10^{-9}$
Metallized	×	<del> -</del>	Kanton		Gold	10,000	.25 (.00635)	5.91 × 10 <sup>-7</sup>	2.9 × 10 <sup>-12</sup>
Films (2)	×		Kapton	.50 (.0127)	Gold	5,000	.50 (.0127)	9.1 × 10-6	9.0 × 10-11
(Sputtered	×		Kapton		Gold	10,000		$7.3 \times 10^{-6}$	$7.2 \times 10^{-11}$
one side only)	×		Kapton		Aluminum	10,000	- ·	1.92 × 10°0	1.90 × 10-11
	×							2.75 × 10-0	2.73 × 10 0 00 × 10-11
	< >							2.23 × 10 <sup>-6</sup>	$2.21 \times 10^{-11}$
	< ×	-			1-7			2.06 × 10 <sup>-6</sup>	$2.04 \times 10^{-11}$
	 {	×		•				1.70 × 10-6	1.68 × 10-11
		×						$5.72 \times 10^{-6}$	5.66 × 10-11
		×						$1.29 \times 10^{-3}$	$1.28 \times 10^{-10}$
		×	-		<b>-</b>	-	10.0	3.55 × 10 <sup>-6</sup>	3.51 × 10-11
		×	Kapton	.50 (.0127)	Aluminum	000,00	(7210.) 05.	4.51 × 10 <sup>-5</sup>	4.4× 10
		-					-		

TABLE XV SPECIFIC PERMEABILITY RATES OF PLAIN AND METALLIZED FILMS (Continued)

Thickness   Thic	-		1			Total	E :	
.50 (.0127) Aluminum 10,000 .50 (.0127) 1.08 × 10 <sup>-5</sup> 6.09 × 10 <sup>-6</sup> 6.00	Stress 0% 20%	Material	Thickness mils (mm)	Material	Thickness,	Thi ckness, mils (mm)	Leak Rate cc/sec	Specific Permeability cc-mm/cm2sec.cm Hg
7.48 × 10°6 7.04 × 10°6 6.09 × 10°6 6.09 × 10°6 6.09 × 10°6 1.14 × 10°5 1.05 (.00635) Aluminum 5,000 .25 (.00635) 8.65 × 10°6 1.05 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10°8 1.06 × 10°8 1.07 × 10°8 1.08 × 10°8 1.09 × 10°8 1.09 × 10°8 1.09 × 10°8 1.09 × 10°8 1.09 × 10°8 1.09 × 10°8 1.09 × 10°8 1.09 × 10°8		Mylar	.50 (.0127)	Aluminum	10,000	.50 (.0127)	1.08 × 10-5	1.07 × 10-10
25 (.00635) Aluminum 5,000 .25 (.00635) 5.13 × 10 <sup>-6</sup> 25 (.00635) Aluminum 5,000 .25 (.00635) 5.13 × 10 <sup>-6</sup> 25 (.00635) Aluminum 5,000 .25 (.00635) 8.65 × 10 <sup>-6</sup> 25 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup> 25 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup> 25 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup>							7.48 × 10 <sup>-6</sup>	7.41 × 10-11 4 o7 × 10-11
25 (.00635) Aluminum 5,000 .25 (.00635) 5.13 × 10 <sup>-6</sup> 6.09 × 10 <sup>-6</sup> 7.25 (.00635) Aluminum 5,000 .25 (.00635) 8.65 × 10 <sup>-6</sup> 6.45 × 10 <sup>-6</sup> 6.45 × 10 <sup>-6</sup> 7.55 × 10 <sup>-6</sup> 7.50 (.0127) 7.50 × 10 <sup>-6</sup> 7.50							0 × 40. /	0.7/× 10 / 03 10-11
.25 (.00635) Aluminum 5,000 .25 (.00635) 5.13 × 10 <sup>-6</sup> .25 (.00635) Aluminum 5,000 .25 (.00635) 5.13 × 10 <sup>-6</sup> .25 (.00635) Aluminum 5,000 .25 (.00635) 6.45 × 10 <sup>-6</sup> .25 (.00635) Aluminum 5,000 .25 (.00635) 8.65 × 10 <sup>-6</sup> .25 (.00635) Gold 5,000 .25 (.0127) 4.33 × 10 <sup>-8</sup> .25 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup> .25 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup> .25 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup>							6.09 × 10 <sup>-6</sup>	6.03 × 10 1 52 × 10-11
.25 (.00635) Aluminum 5,000 .25 (.00635) 5.13 × 10 <sup>-6</sup> 6.09 × 10 <sup>-6</sup> 7.55 × 10 <sup>-6</sup> 8.69 × 10 <sup>-6</sup> 8.69 × 10 <sup>-6</sup> 8.69 × 10 <sup>-6</sup> 1.32 × 10 <sup>-6</sup> 1.32 × 10 <sup>-6</sup> 1.32 × 10 <sup>-6</sup> 1.32 × 10 <sup>-6</sup> 6.45 × 10 <sup>-6</sup> 6.45 × 10 <sup>-6</sup> 7.55 × 10 <sup>-6</sup> 6.45 × 10 <sup>-6</sup> 6.4	>						1 14 × 10 <sup>-5</sup>	1.13 × 10 <sup>-10</sup>
.50 (.0127) Aluminum 10,000 .50 (.0127) 5.64 × 10 <sup>-6</sup> .50 (.0127) Aluminum 5,000 .25 (.00635) 5.13 × 10 <sup>-6</sup> .25 (.00635) Aluminum 5,000 .25 (.00635) 8.65 × 10 <sup>-6</sup> .25 (.00635) Aluminum 5,000 .25 (.00635) 8.65 × 10 <sup>-6</sup> .25 (.00635) Gold 5,000 .50 (.0127) 4.33 × 10 <sup>-8</sup> .25 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup> 1.04 × 10 <sup>-8</sup> 1.05 (.00635) 3.89 × 10 <sup>-8</sup>	< >						5 04 × 10-6	11-01 28 2
.25 (.00635) Aluminum 5,000 .50 (.0127) 5.64 × 10 <sup>-6</sup> .25 (.00635) Aluminum 5,000 .25 (.00635) 5.13 × 10 <sup>-6</sup> .25 (.00635) Aluminum 5,000 .25 (.00635) 8.65 × 10 <sup>-6</sup> .25 (.00635) Gold 5,000 .50 (.0127) 4.33 × 10 <sup>-8</sup> .25 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup> .25 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup>	< ×						1.04 × 10 <sup>-5</sup>	7.26 × 10 <sup>-10</sup>
.50 (.0127) Aluminum 10,000 .50 (.0127) 5.64 × 10 <sup>-6</sup> .25 (.00635) Aluminum 5,000 .25 (.00635) 5.13 × 10 <sup>-6</sup> 4.22 × 10 <sup>-6</sup> 8.69 × 10 <sup>-6</sup> 8.69 × 10 <sup>-6</sup> 1.32 × 10 <sup>-6</sup> 1.32 × 10 <sup>-6</sup> 1.32 × 10 <sup>-6</sup> 25 (.00635) Aluminum 5,000 .25 (.00635) 8.65 × 10 <sup>-6</sup> 25 (.00635) Gold 5,000 .50 (.0127) 4.33 × 10 <sup>-8</sup> 5.25 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup> 1.04 × 10 <sup>-8</sup> 1.04 × 10 <sup>-8</sup> 1.05 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup>	×	 -	-	•	•	-	6.09 × 10-6	6.03 × 10 <sup>-11</sup>
.25 (.00635) Aluminum 5,000 .25 (.00635) 5.13 × 10 <sup>-6</sup> 2.51 × 10 <sup>-6</sup> 4.22 × 10 <sup>-6</sup> 8.69 × 10 <sup>-6</sup> 1.32 × 10 <sup>-6</sup> 1.32 × 10 <sup>-6</sup> 1.32 × 10 <sup>-6</sup> 2.5 (.00635) Aluminum 5,000 .25 (.00635) 8.65 × 10 <sup>-6</sup> 3.00 × 10 <sup>-6</sup> 5.000 3.50 (.0127) 4.33 × 10 <sup>-8</sup> 5.10 × 10 <sup>-8</sup> 1.04 × 10 <sup>-8</sup> 1.25 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup> 1.38 × 10 <sup>-8</sup> 1.39 × 10 <sup>-8</sup>	×	 Mylar	.50 (.0127)	Aluminum	10,000	.50 (.0127)	5.64 × 10 <sup>-6</sup>	5.58 × 10 <sup>-11</sup>
2.51 × 10 <sup>-6</sup> 4.22 × 10 <sup>-6</sup> 8.69 × 10 <sup>-6</sup> 1.32 × 10 <sup>-6</sup> 1.32 × 10 <sup>-6</sup> 1.32 × 10 <sup>-6</sup> 1.32 × 10 <sup>-6</sup> 2.5 (.00635) Aluminum 5,000 .25 (.00635) 8.65 × 10 <sup>-6</sup> 3.00 × 10 <sup>-6</sup> 3.00 × 10 <sup>-6</sup> 3.00 × 10 <sup>-6</sup> 5.25 × 10 <sup>-8</sup> 5.25 × 10 <sup>-8</sup> 5.25 × 10 <sup>-8</sup> 1.04 × 10 <sup>-8</sup> 1.04 × 10 <sup>-8</sup> 1.05 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup>		Kapton	.25 (,00635)		5,000	,25 (,00635)	5.13 × 10-6	2.54 × 10-11
.25 (.00635) Gold 5,000 .25 (.0127) 4.13 × 10 <sup>-8</sup> .25 (.00635) Gold 5,000 .50 (.0127) 4.33 × 10 <sup>-8</sup> .25 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup> .25 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup>		 			. —		$2.51 \times 10^{-6}$	$1.24 \times 10^{-11}$
25 (.00635) Gold 5,000 .50 (.0127) 8.69 × 10 <sup>-6</sup> 3.00 × 10 <sup>-6</sup> 5.25 × 10 <sup>-6</sup> 3.00 × 10 <sup>-6</sup> 5.25 × 10 <sup>-8</sup> 5.25 × 10 <sup>-8</sup> 7.38 × 10 <sup>-8</sup> 7.39 × 10 <sup>-8</sup> 7.30 × 10 <sup>-8</sup> 7.31 × 10 <sup>-8</sup> 7.32 × 10 <sup>-8</sup> 7.33 × 10 <sup>-8</sup> 7.34 × 10 <sup>-8</sup> 7.35 × 10 <sup>-8</sup> 7.35 × 10 <sup>-8</sup> 7.35 × 10 <sup>-8</sup> 7.38 × 10 <sup>-8</sup>		 ·					$4.22 \times 10^{-6}$	2.09 × 10 <sup>-11</sup>
.25 (.00635) Aluminum 5,000 .25 (.00635) 8.65 × 10 <sup>-6</sup> 3.00 × 10 <sup>-6</sup> 3.25 × 10 <sup>-8</sup> 5.25 × 10 <sup>-8</sup> 5.25 × 10 <sup>-8</sup> 7.55 × 10 <sup>-8</sup> 7.55 × 10 <sup>-8</sup> 7.55 × 10 <sup>-8</sup> 7.50 (.0127) 3.89 × 10 <sup>-8</sup> 7.55 × 10 <sup>-8</sup> 7.56 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup>							$8.69 \times 10^{-6}$	4.30 × 10-11
.25 (.00635) Aluminum 5,000 .25 (.00635) 8.65 × 10 <sup>-6</sup> 3.25 (.00635) Gold 5,000 .50 (.0127) 4.33 × 10 <sup>-8</sup> 5.25 × 10 <sup>-8</sup> 7.55 × 10 <sup>-6</sup> 7.55 × 10 <sup>-8</sup> 7.56 × 10 <sup>-8</sup> 7.57 × 10 <sup>-8</sup> 7.58 × 10 <sup>-8</sup> 7.59 × 10 <sup>-8</sup> 7.59 × 10 <sup>-8</sup> 7.59 × 10 <sup>-8</sup> 7.59 × 10 <sup>-8</sup> 7.50 × 1							$1.32 \times 10^{-6}$	6.5 × 10-12
.25 (.00635) Aluminum 5,000 .25 (.00635) 8.65 × 10 <sup>-6</sup> 3.00 × 10 <sup>-6</sup> 8.65 × 10 <sup>-6</sup> 8.50 (.0127) 4.33 × 10 <sup>-8</sup> 5.25 × 10 <sup>-8</sup> 7.00 (.0127) 8.10 × 10 <sup>-8</sup> 1.04 × 10 <sup>-8</sup> 1.05 (.0127) 3.89 × 10 <sup>-8</sup> 1.04 × 10 <sup>-8</sup> 1.05 (.0127) 3.89 × 10 <sup>-8</sup> 1.05 (.0127) 3	×	 <del>,</del>						$2.07 \times 10^{-11}$
.25 (.00635) Aluminum 5,000 .25 (.00635) 8.65 × 10 <sup>-6</sup> .25 (.00635) Gold 5,000 .50 (.0127) 4.33 × 10 <sup>-8</sup> .25 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup> .25 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup>	×	+ 2 Z					$7.55 \times 10^{-9}$	3.74 × 10-11
.25 (.00635) Aluminum 5,000 .25 (.00635) 8.65 × 10 <sup>-6</sup> .25 (.00635) Gold 5,000 .50 (.0127) 4.33 × 10 <sup>-8</sup> .25 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup>	×	 					6.45 × 10-6	3.19 × 10-11
.25 (.00635) Aluminum 5,000 .25 (.00635) 8.65 × 10 <sup>-0</sup> .25 (.00635) Gold 5,000 .50 (.0127) 4.33 × 10 <sup>-8</sup> 5.25 × 10 <sup>-8</sup> 5.10 × 10 <sup>-8</sup> 1.04 × 10 <sup>-8</sup> 1.25 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup>	×	 _	_	-	•	-	$3.00 \times 10^{-6}$	1.49 × 10
.25 (.00635) Gold 5,000 .50 (.0127) 4.33 × 10 <sup>-8</sup> 5.25 × 10 <sup>-8</sup> 5.10 × 10 <sup>-6</sup> 5.10 × 10 <sup>-6</sup> 1.04 × 10 <sup>-8</sup> 1.04 × 10 <sup>-8</sup> 1.25 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup>	×	Kapton	.25 (.00635)	Aluminum	2,000	.25 (.00635)	8.65 × 10 <sup>-6</sup>	4.28 × 10 <sup>-11</sup>
.25 (.00635) Gold 5,000 (.50 (.0127) 3.89 × 10 <sup>-8</sup>		 Kapton	25 ( 00635)		5,000	50 ( 0127)	$4.33 \times 10^{-8}$	4.3 × 10-13
.25 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup>		_	/ccoope) cz.		-	(/7:00)	$5.25 \times 10^{-8}$	$5.2 \times 10^{-13}$
.25 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup>							$5.10 \times 10^{-6}$	5.05 × 10 <sup>-11</sup>
.25 (.00635) Gold 5,000 .50 (.0127) 3.89 × 10 <sup>-8</sup>			-	-	-		1.04 × 10-8	$1.03 \times 10^{-13}$
		 Kapton	.25 (.00635)		2,000	.50(.0127)	3.89 × 10 <sup>-8</sup>	3.84 × 10 <sup>-13</sup>

TABLE XV SPECIFIC PERMEABILITY RATES OF PLAIN AND METALLIZED FILMS (Continued)

	Specific Permeability cc-mm/cm <sup>2</sup> sec-cm Hg	1.85 × 10-11 3.33 × 10-13 5.95 × 10-11 1.16 × 10-10 1.29 × 10-13 6.35 × 10-15 6.35 × 10-12 2.05 × 10-13 2.05 × 10-13 1.40 × 10-13 1.40 × 10-11 5.33 × 10-11	7.59 × 10 <sup>-</sup> 13 1.98 × 10 <sup>-</sup> 15 1.98 × 10 <sup>-</sup> 12 1.94 × 10 <sup>-</sup> 14 5.15 × 10 <sup>-</sup> 13 2.71 × 10 <sup>-</sup> 13 2.71 × 10 <sup>-</sup> 13 5.92 × 10 <sup>-</sup> 13 4.81 × 10 <sup>-</sup> 13 4.75 × 10 <sup>-</sup> 14 2.98 × 10 <sup>-</sup> 14 2.78 × 10 <sup>-</sup> 14 2.78 × 10 <sup>-</sup> 14 7.9 × 10 <sup>-</sup> 15 9.2 × 10 <sup>-</sup> 15
H	Leak Rate cc/sec	1.87 × 10-6 3.37 × 10-8 6.01 × 10-6 1.17 × 10-6 1.87 × 10-8 1.3 × 10-8 5.7 × 10-10 6.42 × 10-6 7.01 × 10-7 2.07 × 10-8 4.51 × 10-8 2.16 × 10-8 2.16 × 10-6 2.13 × 10-6 5.39 × 10-6	8.04 × 10 <sup>-9</sup> 1.0 × 10 <sup>-7</sup> 9.8 × 10 <sup>-10</sup> 2.6 × 10 <sup>-8</sup> 1.37 × 10 <sup>-8</sup> 5.0 × 10 <sup>-9</sup> 2.99 × 10 <sup>-9</sup> 2.43 × 10 <sup>-9</sup> 1.5 × 10 <sup>-9</sup> 1.5 × 10 <sup>-9</sup> 1.6 × 10 <sup>-9</sup>
Total	Thickness, mils (mm)	.50 (.0127)	1.00 (.0254)
Вu	Thickness, A	5,000	5,000
Coating	Material	P loo	<u>B</u>
Film	Thickness mils (mm)	.25 (.00635)	.50 (.0127)
	Material	Kapton Kapton	Kapton Kapton
Film	Stress 0% 20%	××××	××××× ××××××××××××××××××××××××××××××××
	Concept	Laminated Films (2) (Diffusion Bonded)	

TABLE XV SPECIFIC PERMEABILITY RATES OF PLAIN AND METALLIZED FILMS (Continued)

	IAPL	ADLE XV	אברורור זי	יייייייייייייייייייייייייייייייייייייי			STECIFIC FERMICABILITY NATES OF FESTIVA AND WEIGHEIGHT FIND COMMISSION		
	Film	_		Film	Coating	ß	Total Laminate	Helium (1)	
***************************************	Stres	200%	A404.0A	Thickness mile (mm)	Material	Thi ckness,	Thickness,	Leak Rage	Specific Permeability
Concepi	200	° 07	Maleijai	(11111)	יאומופוומו		(mm) 2	225 /22	71-
Laminațed		×	Kapton	.50 (.0127)	Cold	10,000	1.00 (,0254)	$4.95 \times 10^{-9}$	9.8 × 10 13
Films (2)		×	<del></del> -					$3.05 \times 10^{-8}$	6.04 × 10 13
(Dirrusion Ronded)		< ×		->		-	-	4 71 × 10 0	$9.33 \times 10^{-14}$
(200		×	Kapton	.50 (.0127)	Gold	10,000	1.00 (.0254)	1.93 × 10 <sup>-0</sup>	$3.82 \times 10^{-13}$
Laminated Film (4)	×		Kapton	.25 (.00635)	Gold	10,000	.50 (.0127)	$1.16 \times 10^{-5}$	$1.15 \times 10^{-10}$
(Vacuum Deposited-	×		_	_				$2.17 \times 10^{-3}$	$2.15 \times 10^{-12}$
Diffusion Bonded)	×							9.76 × 10.8	$\frac{9.66 \times 10^{-13}}{10^{-13}}$
	×								: <u>×</u>
	×						-	5 × 2	$4.71 \times 10_{-11}$
		×						8.4 × 10 1, 1, 10-5	8.31 × 10 10
		××						1.6/× 10_5	01 × 02 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
	-	××						2	
		×	-	-	-	-	-	× Z	
	×		Kapton	.50 (.00635)	Gold	10,000	1.00 (.0254)	$3.02 \times 10^{-2}$	5.98 × 10 11
	× >				-			2.76 × 10 <sup>-6</sup>	
	< ×				_			1.94 × 10 <sup>-6</sup>	3.84 × 10-11
	×				-			7.01 × 10.7	â
		×			_			$4.7 \times 10^{-5}$	9.3 × 10 10
		×							$\begin{bmatrix} 2.38 \times 10_{10} \\ 1.38 \times 10_{10} \end{bmatrix}$
		× >			<u> </u>			8.21 × 10 9.87 × 10	1.85 × 10 10
		<×		·····•	<b>,</b>			7.67 × 10-6	$1.52 \times 10^{-10}$
				<b>-</b>	-	-	-		
(1)				2.	(3)				

(1) Based on a  $t_{\rm CS}$  area of  $38.5~{\rm in}^2~(248.3~{\rm cm}^2)$ . (2) Sputtered gold and aluminum coatings.

(3) Plain film values are average readings taker from Task 1. (4) Vacuum deposited gold coatings.

TABLE YVI-SUMMARY - AVERAGE SPECIFIC PERMEABILITY RATES AT +70°F (21°C)

	17	ABLE 71	WDS-1	MARY – AVEKI	AGE SPECIFI	C PERMEABI	TABLE ZVI-SUMMARY - AVERAGE SPECIFIC PERMEABILITY RATES AT +70°F (21°C)	19 F (21°C)
		F:1 <sub>m</sub>	-	F:1 p.	Coating	ina	Total	Specific Parmoohility
		= (				,	Specimen	Andream is a consider
Concept	Material	5tress 0%	20%	lhickness, mils (mm)	Material	Ihickness.	Thickness mils (mm)	ot Heliym (cc-mm/cm²sec.cm Hg)
Plain Film	Mylar	×		.25 (.00635)		1	.25 (.00635)	$11.4 \times 10^{-9}$
		,	×	_	ı	1	$\overline{}$	54.8 × 10 <sup>-9</sup>
		×		_	ı	ı	$\overline{}$	$0.24 \times 10^{-7}$
			×	_	1	١.	$\overline{}$	16.8 × 10 <sup>-7</sup> ,
	Kapton	×		$\sim$	· ·	1	.25 (.00635)	$3.73 \times 10^{-4}$
			×	_	•	ı		95.0 × 10 <sup>-7</sup> ,
		×			ı	ı		$0.40 \times 10^{-3}$
			×	_	ı	ı	$\overline{}$	$1.73 \times 10^{-3}$
		×		1,00 (,0254)	<b>!</b>	l	1.00 (.0254)	$2.49 \times 10^{-7}$
Metallized	Kapton	×		.25 (00635)	Ccld	10,000	_	
(Sputtered One	•	×		_	=	5,000	_	
Side Orly)		×		_	Gold	10,000	$\overline{}$	× 10- :
		×		_	Altminum	10,000	$\overline{}$	5 × 10 11
			×	_	=	.=	$\overline{}$	$5.72 \times 10^{-11}$
	Mylar	×		/ ~	:	=	.50 (.0127)	7, 13 × 10,
			×		-	10,000	.50 (.5127)	$20.3 \times 10^{-11}$
	Kapton	×			=	5,000	.25 (.00635)	$2.1 \times 10^{-11}$
			×	.25 (.00635)	Aluninum	5,000	.25 (.006°5)	$2.95 \times 10^{-11}$
Laminated Film	Kapton	×		.25 (.60635)	Gold	5,000	.50 (.0127)	$3.59 \times 10^{-13}$
(Sputtered Coating-	•		×	/ \ <u> </u>	<u>-</u>	5,000		426.0 × 10
Diffusion Bonded)		×			-	10,000	.50 (.0127)	
			×	_	3	10,000	$\overline{}$	$139.0 \times 10^{-13}$
		×		•	-	5,000	_	×
			×	$\overline{}$	-	5,000	~ ·	$3.61 \times 10^{-13}$
		×			-	10, 000		$0.73 \times 10^{-13}$
			×	.50 (.0127)	Gold	10, coo	1,00 (.0254)	$2.65 \times 10^{-13}$
Laminated Film	Kapton	×		.25 (.05635)	Gold	10,000	$\overline{}$	$7.7 \times 10^{-11}$
(Vacuum Deposited			×	.25 (.00655)	÷	10,000	.50 (.0127)	×
Coating - Diffusion		×		$\overline{}$	-	10,000	1.00 (.0254)	χ
Bonding)			×	.50 (.0127,	-	10,000	1,00 (,0254)	16.8 × 10-11
(1) Single Values	Values	_	•	-	•	•	•	

(1) Single Values

The test specimens can be classified into four categories: (1) plain film, (2) film metallized one-side only, (3) sputtered coating followed by diffusion bonding, and (4) vacuum deposited coating followed by diffusion bonding. Each of these materials are discussed in the following paragraphs.

- a. <u>Plain Film</u> The plain Mylar and Kapton films were tested primarily to serve as control with most of the data coming from Task I, Section 3.1. Restating some of the conclusions from Section 3.1: the Kapton film is less permeable to helium gas than the Mylar films of comparable thickness. In both types of films the 0.25 mil thick film is significantly more permeable and is less uniform than a 0.50 mil film. Stressing the plain film (20% biaxial) increases the helium permeability by 1 to 3 times over the non-stressed film.
- b. Metallized Film (one side only) Samples of Mylar and Kapton were sputter-coated with gold and aluminum to assess the effectiveness of such a coating to reduce permeability. Tests have demonstrated (Refererences 4 and 6 ) that vacuum metallizing alone is not an effective means of reducing the gas permeability rate through a thin polymeric film. Sputtering on the other hand, produces a more dense and uniform coating and it was anticipated that reduced permeability would be realized. The results show that a single sputtered coating does reduce the helium permeability by approximately 2 orders of magnitude over the plain film with no significant difference being noted between a gold or aluminum coating. In addition to reducing the permeability of the film, the difference in permeability rates between the stressed and non-stressed film was reduced slightly.
- c. <u>Laminated Film (Sputtered Coating Diffusion Bonded)</u> Diffusion bonding of sputtered coatings reduces the film helium permeability by at least four magnitudes over the plain film and appears independent of the thickness of the metallized coating.

As stated earlier, the plain 1/2 mil thick films have a slightly lower permeability rate than the 1/4 mil thick films due to improved uniformity and this is still evident in the metallized and diffusion bonded film specimens made from the two film thicknesses.

Applying a 20% biaxial stress to the films increases the helium permeability of each specimen type, whether plain, metallized or diffusion bonded. Stressing a diffusion bonded specimen does not delaminate or appear to deteriorate the diffusion bond but in some cases increased the helium permeability by two orders of magnitude over the unstressed film. Even so, the stressed diffusion bonded specimens had a lower permeability than the unstressed metallized film.

Laminated Film (Vacuum Deposited Coating - Diffusion Bonded) - Diffusion bonding films having a vacuum deposited coating rather than a sputtered coating is not an effective method of reducing permeability. The helium permeability of a diffusion bonding of film with vacuum deposited coating is in the range of  $10^{-11}$  cc-mm/cm<sup>2</sup> sec cm Hg, which is the same magnitude as the unbonded film with a sputtered coating on one side. Consequently, there is little advantage to diffusion bonding two films when the same results can be obtained by merely applying a sputtered coating rather than a vacuum deposited coating. However, one important trend should be noted from these results. A thin polymeric film with a vacuum deposited coating on one side has a helium permeability rate in the same range as that of plain film or  $10^{-9}$  cc-mm/cm<sup>2</sup> sec. cm Hg. Diffusion bonding the vacuum metallized film reduces the permeability by 2 orders of magnitude or to  $10^{-11}$  cc-mm/cm<sup>2</sup> sec. cm Hg. Similarly, diffusion bonding a film with a sputtered metallized coating reduces the permeability from  $10^{-11}$  to  $10^{-13}$  or 2 orders of magnitude. This demonstrates diffusion bonding two metallized films together is an effective method of reducing gas permeation but the degree of reduction is highly dependent upon the quality of the metallized coatings. Evidently the porosity of the vacuum deposited coating was sufficiently high that the pore could not be effectively sealed.

### d. (cont.)

The thickness of the metallized coating, within the range tested, had no significant effect on the permeability. That is, the  $5000\text{\AA}$  coatings were just as effective as the  $10,000\text{\AA}$  coating, indicating that the break-off point is somewhat less than  $5,000\text{\AA}$ .

### 3.2.3 Film Flexibility

Twist-flex specimens were prepared from each of the 10 materials shown in Table XIV. In addition, several control samples were tested to provide data comparisons. The specimens and test procedures were the same as those described in Section 3.1.2. The data is presented in Table XVII. All testing was conducted at -320°F (-195°C).

The presence of a metallized surface on the polymeric films, either sputtered or vacuum deposited and whether or not it has been diffusion bonded into a laminate, reduces the Twist-flex life of the film. Diffusion bonding itself, however, does not appear to influence the Twist-flex life. For example, 1/4 mil Kapton laminate with  $10,000\text{\AA}$  of sputtered gold (total laminate thickness = 1/2 mil) had an endurance limit of 50-60 cycles, while an unlaminated sample of the same metallized film (total thickness = 1/4 mil) went less than 75 cycles. The diffusion bonded sample being thicker should have had a substantially lower threshold, but this was not the case. A similar comparison was obtained with the 1/2 mil Kapton film. A similar trend seemed to be present with the vacuum-metallized diffusion-bonded specimens, but the data scatter makes it difficult to assess.

The 1/4 mil film samples sputtered with aluminum had a lower threshold than the same film sputtered with gold. The 1/4 mil Kapton film sputtered with aluminum had a damage threshold at about 40 cycles compared to a threshold value of 75 cycles for sputtered gold on 1/4 mil Kapton. Comparing the results shown in Table XVII, it appears that both metallizing processes degrade the twist-flex life with the vacuum metallizing, causing more scatter than sputtering, although the latter process causes more degradation.

TABLE XVII-TWIST-FLEX RESULTS AT -320°F (-195°C)

A         mils (mm)         Cycles         Results         (cc/sec)           -         1.0 (.0254)         80         Torn         -           -         1.0 (.0254)         65         Torn         -           -         1.0 (.0254)         60         Torn         -           -         1.0 (.0254)         90         Intact         -           -         1.0 (.0254)         90         Intact         -           -         2 (.0127)         125         Intact         -           -         25 (.0127)         120         Intact         -           2,000         .25 (.00635)         25         Intact         -           5,000         .25 (.00635)         25         Intact         -           5,000         .25 (.00635)         35         Intact         -           10,000         .25 (.00635)         35         Intact         -           5,000         .25 (.00635)         35         Intact         -           10,000         .3 (.0127)         25         Ion         -           10,000         .5 (.0127)         3         Intact           10,000         .5 (.0127)         3
1.0 (.0254) 80 Torn 1.0 (.0254) 65 Torn 1.0 (.0254) 90 Intact 1.0 (.0254) 90 Intact 1.0 (.0254) 90 Intact 25 (.0127) 225 Torn 25 (.0127) 25 Torn 25 (.00635) 25 Intact 45 Intact 46 Intact 47 Small tear 5 (.0127) 25 Torn 5 (.0127) 25 Torn 6 Torn 7 Small tear
1.0 (.0254) 60
1.0 (.0254) 90 Intact .5 (.0127) 125 Intact 225 Torn .5 (.0127) 5 Torn .25 (.00635) 75 Torn .25 (.00635) 25 Intact .25 (.00635) 25 Intact 40 Intact 45 Small tear 13 .25 (.00635) 35 Small tear 15 (.0127) 25 Torn 5 Small tear 5 (.0127) 25 Small tear 5 Small tear
.5 (.0127) 125 Intact 225 Torn .5 (.0127) 5 Torn (2) .25 (.00635) 75 Torn 140 Torn .25 (.00635) 25 Intact 40 Intact 40 Intact 45 Small tear 15 (.0127) 25 Torn 5 (.0127) 25 Torn 5 (.0127) 25 Small tear 5 (.0127) 25 Small tear 5 Small tear
.5 (.0127) 5 Torn (2) .25 (.00635) 75 Torn .5 (.0127) 120 Intact .25 (.00635) 25 Intact 40 Intact 45 Small tear 45 Small tear 10 Torn 5 (.0127) 25 Small tear 5 Small tear 7 Small tear
.5 (.0127) 5 Torn (2) .25 (.00635) 75 Torn .5 (.0127) 120 Intact 140 Torn .25 (.00635) 25 Intact 40 Intact 445 Small tear 45 (.0127) 25 Torn 5 (.0127) 25 Torn 5 Small tear 5 (.0127) 25 Small tear 5 Small tear
.25 (.00635) 75 Forn .5 (.0127) 120 Intact .25 (.00635) 25 Intact 40 Intact 45 Small tear 45 " " " .25 (.00635) 35 Small tear 5 (.0127) 25 Small tear 5 (.0127) 5 Small tear 7 Small tear
.5 (.0127) 120 Intact 140 Torn  .25 (.00635) 25 Intact 40 Intact 45 Small tear 45 " " "  .25 (.00635) 35 Small tear 5 (.0127) 25 Torn 5 Small tear
.25 (.00635) 25 Intact 50 Forn 40 Intact 45 Small tear 45 " " " 46 " " " 5 (.0127) 25 Forn 5 Small tear 5 (.0127) 25 Forn 5 Small tear
.25 (.00635) 25 Intact 50 Torn 40 Intact 45 Small tear 45 " " " 40 " " " 5 (.0127) 25 Torn 5 (.0127) 25 Small tear 5 Small tear
50 Torn 40 Intact 45 Small tear 45 " " " 40 " " " 5 (.0127) 25 Torn 5 (.0127) 5 Small tear
40 Intact 45 Small tear 45 1
45 Small tear 45 " " 40 " " 5 (.006.35) 35 Small tear  .5 (.0127) 25 Torn 5 Small tear
.25 (.006.35) 35 Small tear 1 .5 (.0127) 25 Forn 5 Small tear 5 Small
.25 (.006.35) 35 Small tear 1 .5 (.0127) 25 Torn 5 Small tear 5 Small
.5 (.006.35) 35 Small tear 1 .5 (.0127) 25 Torn 5 Small tear 5
.5 (.0127) 25 Torn 10 Torn 5 Small tear 5 (.0127) 3 Intact 5 Small tear
10 Torn 5 Forn 5 Small tear 7 5 (.0127) 3 Intact 5 Small tear 5 Small tear
5 Forn 5 Small tear >2 6 (.0127) 3 Intact 5 Small tear
5 Small tear >2 5 (.0127) 3 Intact 5 Small tear
.5 (.0127) 3 Intact 5 Small tear
.5 (.0127) 3 Intact 5 Small tear
Small tear

TABLE XVII TWIST-FLEX RESULTS AT -320°F (-195°C) (Continued

						Total			
			Film	ပိ	Coating	Specimen			Leak Rate
Concept	Specimen No.	Film	Thickness mils (mm)	Material	Thickness A	Thickness mils (mm)	Cycles	Results	After Test (cc/sec)
Metallized	T1/4M5KAAL-1	Mylar	.25 (.00635)	Aluminum	5,000	.25 (.00635)	15	Intact	,
Film (	!	_	-				25	Torn	ı
(Sputtered one	-2						15	Intact	ı
side only)						-	25	Small tear	1.8
	£-						40	Intact	ı
							20	Torn	86.0
	4-	<b>-</b>	-	<b>&gt;</b>	-	-	40	Small tear	6.7
	-5	Mylar	.25 (.00635)	Aluminum	2,000	.25 (.00635)	90	Intact	8.0
	1 48/01/10/11	-	F010 / 03		000 01	icio y	u	100	!
	11/2MIUKAAL-1	Mylar	(/710.) 00.		000.	(/710.) 00.	٠ ٢	1000	I
			-		-		2 '	lorn	ı
	-2						2	Torn	ı
	-3					-	2	Small tear	18.8
	-4						2	Intact	ı
							15	Small tear	15.3
	-5	_	<b>&gt;</b>	<b>&gt;</b>	_	-	2	Small tear	1
	9-	Mylar	.50 (.0127)	Aluminum	10,000	.50 (.0127)	5	Intact	23.5
Metallized	T1/4KV10KAAu-1	Kapton	.25 (.00635)	Gold	10,000	.25 (,00635)	35	Intact	ı
Film (5)		-	-	-	<b>*</b>	•	45		300 -\-
(Vacuum	-2	Kapton	.25 (.00635)	Gold	10,000	.25 (.00635)	31	Torn	<b>~</b> 500
Deposited one									
y and and	T1/2KV10KÅAu-1	Kapton	.50 (.0127)	Gold	10,000	.50 (.0127)	99	Intact	ı
	-2	Kapton	50 ( 0127)	<u>- 1</u> €	10.000	50 ( 0127)	80 40	Torn	1 1
				3		(/7.0.) 00.			

TABLE XVII TWIST-FLEX RESULTS AT -320°F (-195°C) (Continued)

Leak Rate	After Test (cc/sec)		1	1	0.23	2.61	ı	ı	200.0	15.0	ı	ı	65.0	2.1	150.0	$7.14 \times 10^{-4}$	6.4	128.0	59.2	33.9	42.3	1		ı	ı	21.0	5.6	ı	0.09		14.8
		Torn	Torn (2)		Intact	Intact (Inflation)	Torn ②	Torn	Small hole	Intact(Inflation)	Torn	Torn	Small hole	Tear, one ply	Small tear	Intact	Intact	3 tears	2 tears	Small tear	Small tear	Torn	Intact	Torn	Intact	Small tear	Small tear	Intact	Intact	Intact	Intact
	-	دراده 06		8	85	75	20	75	2	9	65	25	2	5	9	4	4	01	5	•	5	01	15	25	5	0	0	2	12	25	30
Total Specimen	Thickness mils (mm)	.50 (.0127)								-	.50 (.0127)	1.00 (.0254)				-	1,00 (.0254)	1.00 (.0254)	,		-	1.00 (.0254)	.50 (.0127)					-			<u> </u>
Coating	Thickness 8	5.000	_			5, 000	10,000	_		-	10,000	2,000	_			-	2,000	10,000			-	10,000	10,000								<b>-</b>
Š	Material	Gold	_							· ~	Pool Cold	Gold				~	Gold	Gold		<u> </u>	-	Gold	Gold							->	<b>~</b>
Film	Thickness mils (mm)	.25 (.00635)	_								.25 (,00635)	.50 (.0127)				-	.50 (.0127)	.50 (.0127)			-	.50 (.0127)	.25 (.00635)							->	<u>-</u>
	Fi E	Kapton			····		· <del>····</del>	<del></del> .		-	Kapton	Kapton				-	Kapton	Kapton	-		<b>-</b>	Kapton	Kapton							->	<b>b</b>
	Specimen No.	T1/4KSGS-1	-2	ဇှ	4-	-5	T1/4K10GS-1	-2	۳-	4-	-5	T1/2K5GS-1	-2	ကု	4-	-5	9-	T1/2K10GS-1	2	ကု	4-	-5	T1/4K10GV-1		-2		ကို	4-	ı	-5	-
	Concept	Laminated	Films -	eq	Coating-		Bonded) (4)																Laminated	Films (5)	(Vacuum	Deposited-		Bonded) (4)			-

TABLE XVII TWIST-FLEX RESULTS AT -320°F (-195°C) (Continued)

Leak Rate	(cc/sec)	1	ı	1	ſ	ı	ı	<b>↑</b> 200	ı	
	Results	Intact	Torn	Torn	Intact	Torn	Intact	Small tear	Intact	Torn
	Cycles	30	40	. 10	10	20	10	15	10	15
Total Specimen	Thickness mils (mm)	1.00 (.0254)								
Coating	Thickness Å	10,000								-
ු <u>ප</u>	Material	Gold								•
Fi	Thickness mils (mm)	.50 (.0127)								-
	Film	Kapton	_							•
	Specimen No.	T1/2K10GV-1		-2	£-		-4	- <del></del>	-5	
	Concept	Laminated	Films (5)	(Vacuum	Deposited -	Diffusion	Bonded) (4)	(Cont'd)		

Coating applied by sputtering.

Bushing freeze-up causing shaft to bind causes tear from short side to long side or the shearing off of the taped edge.

1000 Å gold both sides vacuum deposited.

Thickness of diffusion bonded specimens is twice the base film thickness.

Coating applied by vacuum deposition.  $\bigcirc$   $\bigcirc$   $\bigcirc$   $\bigcirc$   $\bigcirc$   $\bigcirc$   $\bigcirc$  However, in applying the aluminum coating, a higher particle energy level was required in the sputtering operation to promote good adherence of the aluminum and could account for the greater degradation.

All the metallized films had a lower threshold value than the plain films which indicated that metallizing by either process (i.e. vacuum-deposition or sputtering) deteriorates the flexibility of the film. Therefore, in order to reduce film permeability it appears that a sacrifice in film flexibility will result. It is significant to note in Figure 25 that metallized diffusion bonded Kapton film, even though it is degraded by metallization has a twist-flex life which is as good as Mylar film of the same thickness.

# 3.2.4 Bond Strength Determination

Butt and lap joint specimens were prepared from each of the metallized and diffusion bonded film concepts using the basic joint configurations given in Section 3.1.3. GT-100 and GT-300 adhesives were again used. For the sputtered aluminum specimens, the lap joints were prepared by bonding the two metallized surfaces together with GT-100. These were the only samples in which the primary bond was between two metallized coatings. The butt joints on the metallized films had one GT-300 doubler bonded to the metallized surface and a second bonded to the plain film side. The joint concepts are shown in Figure 18.

The results of the bond tests are shown in Table XVIII. The predominant mode of failure in all samples, with the exception of the B1/2KSB1OKAAu and L1/2KSB1OKAAu series, was in the film itself. The mode of failure of the B1/2KSB1OKAAu and L1/2KSB1OKAAu series (diffusion bonded 10,000A gold on 1/2 mil Kapton) were termed joint failures since the films separated in the joint region at the diffusion bonded layer. The stresses in the joints of these specimens were considerably higher than that in other samples which accounts for the failure of the diffusion bond. The tests were conclusive in showing that very high stresses can be developed in the films prior to bond failure and that conventional bladder fabrication techniques would be applicable.

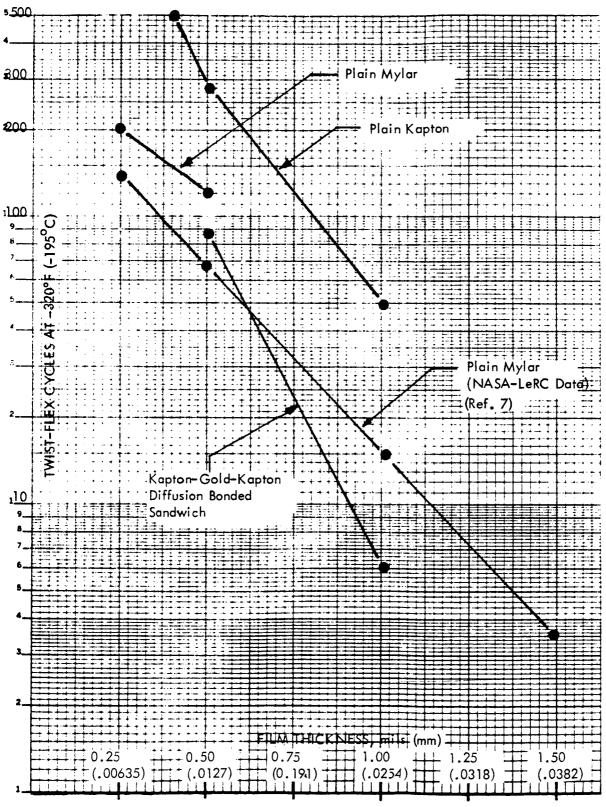


FIGURE 25 - RELATIONSHIP OF FILM THICKNESS TO TWIST-FLEX CYCLIC LIFE AT -320°F (-195°C)

TABLE XVIII BOND STRENGTH TENSILE TESTS - +70°F (21°C)

So No.	Gage Length,	Load Rate, Lap,	Lap,	Width,	Load,	Film Stress, Ksi (Ka/Sa.cm)	Lap Stress, Psi (Ka/Sa.cm)	Failure
.25 mil (.00635 mm) Kapton sputtered with 5000 A Aluminum on one side.	Kapton sputters	sd with 5000 Å A	luminum on or	ne side.		BUTT JOINT		
81/4K5KÅA <b>L</b> -1 -2 -3 -4	10.0(25.4)	2.0 (5.08)	1/2*(1.27)*	.54)	8.75 (3.99) 5.00 (2.27) 6.25 (1.29) 3.25 (1.47) 6.25 (1.29)	35.0 (2460.7) 20.0 (1406.1) 25.0 (1757.6) 13.0 (914.0) 25.0 (1757.6)	8.25 (.58) 5.00 (.35) 6.25 (.44) 3.25 (.23) 6.25 (.44)	Film Film Adjacent Film
L1/4K5KÅA <b>L</b> -1 -2 -3 -4 -4	10.0 (25.4)	1.0 (2.54)	1/2 (1.27)	1.0 (2.54)	4.75 (2.15) 5.50 (2.50) 6.25 (1.29) 4.75 (2.15) 5.75 (2.61)	LAP JOINT 19.0 (1335.9) 22.0 (1546.7) 25.0 (1757.6) 19.0 (1335.9) 23.0 (1617.1)	9.50 (.67) 11.00 (.77) 12.50 (.88) 9.50 (.67) 11.50 (.81)	Film Film Adjacent Film Film
.50 mil (.0127 mm) Kapton sputtered with 10,000 Å Aluminum on one side  B1/2K10KÅA <b>L</b> -1 8.0 (20.3) 1.0 (2.54) 1/2* (1.27)* 1.0 (2.54) -3 -4	Kapton sputtered	1.0 (2.54)	Aluminum on o	.54)	11.00 (4.99) 10.75 (4.88) 11.75 (5.33) 9.75 (4.42) 10.75 (4.88)	BUTT JOINT 22.0 (1546.7) 21.5 (1511.6) 23.5 (1652.2) 19.5 (1371.0) 21.5 (1511.6)	11.00 (.77) 10.75 (.76) 11.75 (.8°) 9.75 (.68) 10.75 (.76)	Fila Fila Grip
L1/2K10KÅA <b>L</b> -1 -2 -3 -4 -4	8.0 (20.3)	1.0 (2.54)	1/2 (1.27)	1.0 (2.54)	1.0 (2.54)12.25 (5.56) 11.75 (5.33) 14.00 (6.35) 13.75 (6.24) 11.50 (5.22)	24.5 (1722.5) 23.5 (1652.2) 28.0 (1968.6) 27.5 (1973.4) 23.0 (1617.1)	24.5 (1.68) 25 (1.62) 28.0 (1.97) 27.5 (1.91) 23.0 (1.61)	Film Film Film Film

\*Each side (overall) lap area = 1 sq. in. (6.45 sq. cm)

TABLE XVIII BOND STRENGTH TENSILE TESTS - +70°F (21°C) (Continued)

	Gage Length,	Load Rate,	Lap,	Width,	Load,	Film Stress, Ksi	Lap Stress,	Failure
25 -: 1 / 00425 1	in. (cm)	in/min(cm/min, in. (cm)	In. (cm)	in. (cm)	lbs (Kg)	(kg/ xq. cm)	(Kg/3q. cm)Location	ocation
.23 mil (.00035 mm) Mylar sputtered with 3000 A Aluminum on one side.	mylar spuriere	a with Jood A A	TO UO MOUIMOI	e side.				
B1/4M5KAA <b>g</b> -1	8.0 (20.3)	1.0 (2.54)	1/2* (1,27)*	1.0 (2.54)	5.50 (2.50)	22.0 (1546.7)	5.50 (.39)	E I
5° -					5.25 (2.38)	21.0 (1476.5)		E E
4 5	<b>→</b>	<b></b>	<b></b>	<b>-</b>	4.50 (2.04) 5.00 (2.27)	18.0 (1265.5) 20.0 (1406.1)		Film Adjacent
						LAP JOINT		
L1/4M5KAA <b>L</b> -1	8.0 (20.3)	1.0 (2.54)	1/2 (1,27)	1.0 (2.54)	4.75 (2.15)			Grip
-3	-				5.25 (2.38)		10.50 (.74)	Film
4-				-	5.50 (2.50)			<u>E</u> :
C-	<b>\</b>	<b>&gt;</b>	•	<b>&gt;</b>	0.20 (2.30)	(/.046.1)	(//-) 00.11	E
.50 mil (.0127 mm) Mylar sputtered with 10,000 A Aluminum on one side.	Mylar sputtered	with 10,000 A A	Vluminum on or	ne side.		BUTT JOINT		
B1/2M10KÅA <b>£</b> -1	8.0 (20.3)	1.0 (2.54)	1/2* (1 .27)*	1 0 (2.54)	13.25 (6.01)	26.5 (1863.1)		Film
7 6	-			13.30 (6.03)	13.30 (6.03)	26.6 (1870.2)	13.30 (.54)	
4-					12.00 (5.44)	24.0 (1687.4)		Adjacent
-5	<b>+</b>	<b>→</b>	<b>&gt;</b>	▶	12.60 (5.72)	25.2 (1771.7)		Film
c						LAP JOINT		
L1/2M10KAA <b>2</b> -1	8.0 (20.3)	1.0 (2.54) 1/2 (1.27)	1/2 (1.27)	1.0 (2.54)	1.0 (2.54) 10.50 (4.77)	23.5 (1652.2)	21.0 (1.48)	Grib Grib
<u>ښ</u>		-			5.90)	26.0 (1828.0)	26.0 (1.83)	Fila
-5	>	-			4.68) 5.24)	23.1 (1624.1)	20.6 (1.45)	Fila
		•		•			(	

\* $E_{\alpha}$ ch side (overall)lap area = 1 sq. in. (6.45 sq. cm).

TABLE XVIII BOND STRENGTH TENSILE TESTS - +70°F (21°C) (Continued)

	Gage	Lond Rate	-8	Width.	700	Film Stress, Ksi	Lap Stress, Psi	, , , , , , , , , , , , , , , , , , ,
Specimen No.	in. (cm)	in/min(cm/min) In. (cm)	In. (cm)	in. (cm)	lbs (Kg)	(Kg/Sq. cm)	(Kg/Sq. cm)	Location
.25 mil (,00635 mm) Kapton vacuum deposited with 10,000 Å Gold on one side and diffusion bonded. Total Film Thickness = .50 mil (.0127 mm)	Kapton vacuur . Total Film T	apton vacuum deposited with 10,000 Å Go Total Film Thickness = .50 mil (.0127 mm	10,000 Å Go	ld on one side		BUTT JOINT		
B1/4KVB10KAAu-1		1.0 (2.54)	1/2 (1.27)	1.0 (2.54)	12.5 (5.69)	25.0 (1757.7)	12.5 (.88)	Film
-5				·	11.0 (4.99)	22.0 (1546.7)	11.0(.77)	Film
۳ ٦					11.0 (4.99)	22.0 (1546.7)	11.0(.7)	ᄪ
4	<b>-</b>	•	-	-	11.0 (4.99)	22.0 (1546.7) 22.0 (1546.7)	11.0(.77)	Film Film
٥						LAP JOINT		
L1/4KVB10KAAu-1	5.5 (13.79)	1.0 (2.54)	1/2(1.27)	1.0 (2.54)	12.0 (5.44)	24.0 (1687.4)		Film
-5					11.6 (5.26)	23.2 (1631,1)	23.2 (	Film
, i					11.4 (5.17)	22.8 (1603.0)		Film
4 '			_	1	9.4 (4.26)	18.8 (1321.8)		Film
<b>5-</b>	<b>&gt;</b>	<b>&gt;</b>	•	>	13.5 (6.12)	27.0 (1898.3)	27.0 (1.90)	Film
.50 mil (.0127 mm) Kapton vacuum deposited with 10,000 A Gold on one side	Sapton vacuum	deposited with 1	0,000 A Gole	on one side	-	BUTT JOINT		
and diffusion bonded. lotal Film thickness = .50 mil (.0127 mm)	. lotal Film	hickness = .50 mi 	1 (.0127 mm)					
B1/2KVB10KAAu-1	5.5 (13.79)	1.0 (2.54)	1/2 (1.27)	1.0 (2.54)	19.8 (8.98)	19.8 (1392.1)	19.8 (1.39)	Film
-5					21.5 (9.75)	21.5 (1511.6)	21.5 (1.51)	Film
					20.2 (9.16)	20.2 (1420.2)	20.2 (1.42)	Film
4 1	->	-			17.0 (7.71)	17.0 (1195.2)	17.0 (1.19)	Film
Ç-	>	•	<b>&gt;</b>	<b>&gt;</b>	24.0 (10.88)	24.0 (1687.4)	24.0 (1.69)	Joint
٥						LAP JOINT		
L1/2KVB10KAAu-1	5.5 (13.79)	1.0 (2.54)	1/2 (1.27)	1.0 (2.54)	23.5 (10.66)	23.5 (1652.2)	47.0 (3.30)	Film
-5					22.0 (9.98)	22.0 (1546.7)	44.0 (3.09)	Joint
7					19.3 (8.75)	19.3 (1356.9)	38.6 (2.71)	Adjacent
4 .		1			22.7 (10.30)	22.7 (1596.0)	45.4 (3.19)	Film
ဂ	•	<b>&gt;</b>	<b>&gt;</b>	<b>&gt;</b>	22.0 (9.98)	22.0 (1546.7)	44.0 (3.00)	Joint
					+			

TABLE XVIII BOND STRENGTH TENSILE TESTS - +70°F (21°C) (Continued)

			•		•			
	Gage	Load Rate	a.	4P!M	, pool	Film Stress, Ksi	Lap Strace. Psi	Failure
Spacimen No.	in. (cm)	in/min(cm/min) In. (cm)	ln. (cm)	in. (cm)	lbs (Kg)	(Kg/Sq. cm)	cm)	Location
.50 (.0127 mm) Kapton sputtered with 5000 and diffusion bonded. Total Film Thickness	ton sputtered wi . Total Film T		A Gold on one side = .50 mil (.0127 mm)			BUTT JOINT		
B1/?KSB5KAAu-1	5.5 (12.79)	1.0 (2.54)	1/2*(1 27)*	1.0 (2.54)	25.0 (11.34)	25.0 (1750.8)	25.0 (1.75)	Joint
-2					25.0 (11.34)	25.0 (1750.8)	25.0 (1.75)	Film
۲,					25.7 (11.66)	25.7 (1799.8)	25.7 (1.80)	Adjacent
4- 7-	<b>→</b>	->	<b>-</b>	<b>→</b>	22.5 (10.21)	22.5 · 1575.7)	22.5 (1.58)	Join? Film
c						LAP JOINT		
L1/2KSB5KAAu-1	5.5 (13.79)	1.0 (2.54)	1/2 (1.27)	1.0 (2.54)	17.0 (7.71)	17.0 (1190.5)	34.0 (2.3%)	Joint
-2						17.0 (1190.5)	34.0 (2.39)	Film
(T)						20.6 (1442.6)	41.2 (2.88)	Joint
4-						21.6 (1512.6)	43.2 (3.04)	Film
-5	<b>&gt;</b>	<b>&gt;</b>	<b>&gt;</b>	<b>&gt;</b>		19.2 (1344.6)	38.4 (2.70)	Joint
,50 mil (.0127 mm) Kapton sputtered with 10,000 A Gold on one and diffusion bended. Total Film Thickness = 1 mil (.0254 mm)	     Capton sputtere   Total Film TF	le d with 10,000 A nickness = 1 mil (	e e	side		BUTT JOINT		
B1/2KSB10KAAu -1	5.5 (13.79)	1.0 (2.54)	1/2*(1.27)*	1.0 (2.54)	23.50 (10.66)	23.50 (1645.7)	23.50 (1.65)	Fila
-2					26.75 (12.13)	26.75 (1873.3)	26.75 (1.87)	
۳	-				15.25 (6.92)	15.25 (1068.0)	15.25 (1.07)	Film
-4	-	>	-	<b></b>	26.75 (12.13) 26.50 (12.02)	26.75 (1873.5) 26.50 (1855.8)	26.75 (1.87) 26.50 (1.36)	Joint
						LAP JOINT		
L1/2KSB10KAAu -1	5.5 (13.79)	1.0 (2.54)	1/2 (1, 27)	1.0 (2.54)	15.50 (7.03)	15.50 (1085.5)	31.00 (2.18)	Joint
-2					13.75 (6.24)	13.75 (962.9)	27.50 (1.93)	Joint
~					19.50 (8.85)	19.50 (1365.6)	39.00 (2.74)	Joint
4 7.					22.75 (10.32)	22.75 (1593.2) 22 50 (1575 7)	45.50 (3.20) 45.00 (3.14)	Joint
>	-	•	•	•	1,300,000,000		0	

\*Each side (overall) lap area = 1 sq. in. (6.45 sq. cm).

TABLE XVIII BOND STRENGTH TENSILE TESTS - +70°F (21°C) (Continued)

ι									
		Gage					Film Stress,	Lap Stress,	;
		Length,	Load Rate,	Lap,	Width,	Load,	Ksi		Failure
	Specimen No.	in. (cm)	in/min(cm/min) In. (cm)	In. (cm)	in. (cm)	lbs (Kg)	(Kg/Sq. cm)	(Kg/Sq. cm) Location	Location
<b></b>	.25 mil (.00635 mm) Kapton sputtered with 5000 Å Gold on one side and	Kapton sputtere	ed with 5000 Å	Gold on one si	de and		BUTT JOINT		
<u>.</u> 0_	diffusion bonded. Total Film Thickness =	otal Film Thickn	less = .50 mil (.0127 mm)	0127 mm)					
	B1/4KSB5KAAu-1	5.5 (13.97)	1.0 (2.54)	1/2*(1.27)*	1.0 (2.54)	15.75 (7.15)	31.5 (2214.7)	15.75 (1.11)	Adjacent
	-2					12.50 (5.67)	25.0 (1757.7)	12.50 (.88)	Adjacent
	က <u>ှ</u>	•				11.50 (5.22)	23.0 (1617.1)	11.50 (.81)	Film (
	4 2		>	<b>→</b>	-	10.75 (4.88) 12.75 (5.78)	25.5 (1792.8)	12.75 (.76)	Grip Grip
<del></del>							LAP JOINT		
	L1/4KSB5K ÅAu-1	5.5 (13.97)	1.0 (2.54)	1/2 (1.27)	1.0 (2.54)	12.75 (5.78)	25.5 (1792.8)	25.5 (1.79)	Film
	-2					10.00 (4.53)	20.0 (1406.1)	20.0 (1.41)	Film
	ဗု	_				11,25 (5,10)	22.5 (1581.9)	22.5 (1.58)	Fi.
72	4-					14.00 (6.35)	28.0 (1968.6)	28.0 (1.97)	Film:
1	-5	<b>&gt;</b>	<b>&gt;</b>	<b>&gt;</b>	<b>&gt;</b>	12.75 (5.78)	25.5 (1792.8)	25.5 (1.79)	Film
• •	.25 mil (.00635 mm) Kapton sputtered with 10,000 A Gold on one side and	Kapton sputtere	d with 10,000 /	A Gold on one	side and		FUTT JOINT		
0	dittusion bonded.   lotal Film   hickness =	otal Film Ihickn _	less = .50 mil (.012/ mm)	J12/ mm)					
-	B1/4KSB10KAAu -1	5.5 (13.97)	1.0 (2.54)	1/2*(1.27)*	1.0 (2.54)	16.50 (7.48)	33.0 (2320.1)	16,50 (1.16)	Film
	ြက္					13.15 (5.96)	26.3 (1849.8)	13.15 (.92)	
	4 "					14.25 (6.46)	28.5 (2003.7)	14.25 (1.00)	Film
		•	•	•	•	02.77	LAP JOINT	0.00	
	-1/4KSB10KAAu -1	5.5 (13.97)	1.0 (2.54)	1/2 (1.27)	1.0 (2.54)	16.50 (7.48)	33.0 (2320.1)	33.0 (2.32)	Adjacent
	-2					15.35 (6.96)	30.7 (2158.4)	30.7 (2.16)	Film
	۳ -					14.25 (6.46)	28.5 (2003.7)	28.5 (2.00)	Film.
	4-		<del></del>			(8/.5) (2/.5)	25.5 (1/92.8)	(5.5 (1.79)	Joint
	?	<b>&gt;</b>	<b>&gt;</b>	<b>&gt;</b>	•	15.00 (6.80)	30.0 (2109.2)	30.0 (2.11)	Film
٠									

\*Each side overall total lap area = 1 sq. in. (6.45 sq. cm).

### 3.3 Advanced Coating Evaluation

The purpose of Task III, Advanced Coating Evaluation, was to select the three most promising metallizing concepts from Task II for more complete evaluation at cryogenic temperatures. Within this phase of the program, film permeability, flexibility, bond strength and interply inflation characteristics were assessed over a temperature range of +25° to -252°C.

#### 3.3.1 Material Selection

Based on the results obtained in Task II (Section 3.2) the following three materials were chosen for more extensive evaluation.

- 1/4 mil Kapton film, sputter coated with 3000A° gold and laminated (Diffusion bonded)
- 1/4 mil Kapton film, sputter coated with 5000A° gold and laminated (Diffusion bonded)
- 3. 1/2 mil Kapton film, sputter coated with 3000A° gold and laminated (Diffusion bonded)

Vacuum deposited coatings were eliminated from the selection since they were not as effective in reducing permeability as sputter-coated films. Vacuum-deposited gold on Kapton diffusion bonded laminates had a helium permeability rate in the order of  $10^{-11}$  cc-mm/cm<sup>2</sup>-sec-cm Hg which is the same level obtained by sputter coating one side of Kapton film. Consequently, there was little advantage to diffusion bonding two films when the same results can be obtained by merely applying a sputtered coating rather than a vacuum deposited coating.

Mylar film, either plain or metallized, was not as impermeable to helium as were the Kapton films of comparable thicknesses and were therefore deleted from the selection. Also, Mylar film is not amenable to the diffusion bonding process; a process which appears to be the most effective method of reducing permeability.

The 3000Å coatings were evaluated in an attempt to increase film flexibility without sacrificing impermeability since it was determined in Task II that the thickness of the metallized coating had little effect on permeability. The selection was also made to aid in establishing the minimum thickness of gold that could be applied and still effectively reduce gaseous permeability. The 5000Å material was selected to obtain data at cryogenic temperatures on the thicker coating.

## 3.3.2 Permeability Measurements

Permeability samples were prepared from each material and tested per the procedures stated in Section 3.1.1. Helium permeability rate determinations were made at  $+25^{\circ}$ C,  $-195^{\circ}$ C and  $-252^{\circ}$ C and liquid hydrogen permeability rates were measured at  $-252^{\circ}$ C. Four replicate samples were run for each data point. The results of the permeability testing for Task III are presented in Table XIX .

All three film materials in Task III had approximately the same order of magnitude of permeability but the 3000Å gold on Kapton (laminated) appears to be slightly more permeable than the 5000Å gold on Kapton (laminated) indicating that 5000Å (10,000Å when laminated) is about the minimum thickness of metallization that can be applied without affecting the gas transmission rate through the film.

The 1/4 mil Kapton with 5000Å gold samples in this task of the program had a higher permeability to helium gas than did the comparable film in Task II (nominally 10<sup>-11</sup> vs 10<sup>-13</sup> cc-mm/cm<sup>2</sup> - sec - cm Hg). Examining the sputtered surfaces revealed that the quality of the metallization in this Task was not as good as that obtained in Task II, Primarily because the sputtering energy was lowered slightly in Task III to minimize film heating. In reducing the sputter — energy the density of the coating was lowered. In Task II it was evident that sputtering damaged the Kapton film slightly due to localized heating. Therefore, this attempt was made to reduce this degradation and increase the flexibility of the film at cryogenic temperatures with little or no sacrifice in permeability. The result was, however, that the permeability dropped to the level obtained with a vacuum metallized coating. The flexibility was increased (see Paragraph 3.3.3).

TABLE XIX SPECIFIC PERMEABILITY RATES OF DIFFUSION BONDED KAPTON FILM LAMINATES

tuatua	Film	Film Stress	Film Thick	Film	Permedia	Test	Gold Coating	Total Laminate	Leak Rate	Specific Permeability
	0%	20%	Mils (mm)	(mm)	Gas	°F(°C)		Mils (mm)	ce/sec	10 11 cc-mm/cm sec.cm Hg
Laminated film (Diffusion Bonded)	××		.25(.	.25 ( .00635)	I	+70 (21) 	3000	.5(.0127)	$6.85 \times 10^{-6}$ $15.60 \times 10^{-6}$	6.78
									•	6.10
	×				<b>-</b> * <b>-</b> .				$4.73 \times 10^{-6}$	4.68
	×								$12.20 \times 10^{-6}$	12.08
		×		-						29.3
		×							•	4.19
		×			· · · · ·				$15.70 \times 10^{-6}$	15.54
		×			-	-	<b></b>	-	1.21 × 10 <sup>-6</sup>	1,19
	×		.25 (.	.00635	Helium	-320 (+95)	3000	.5 (.0127)	$11.90 \times 10^{-6}$	11.78
	×				-	·			13.30 × 10 <sup>-6</sup>	13.17
	×								6.11 × 10-6	6.05
	×					-			5.38 × 10-6	5.33
	×				. *************************************				11.70 × 10 <sup>-6</sup>	11.58
		×							11.80 × 10 <sup>-6</sup>	11,68
		×							1.74 × 10 <sup>-6</sup>	1,72
		×							$12.90 \times 10^{-6}$	12,77
		×	-		-	_	<del>-</del>	-		1.80
	×		.25(.	.00635)	Helium	-423 (-252)	3000	.5(.0127)	1.88 × 10-6	1.86
	×	-	<u>-</u>		_				12.30 × 10-6	12.12
	×									7.95
	×				-				$5.20 \times 10^{-6}$	5.15
-	×		-			_			×	11.38
		×								5.07
•-		×							$2.10 \times 10^{-6}$	2.07
		×							11.1 × 10-6	10.98
•••		×	-		_	-	<b>)-</b> -	-	$3.50 \times 10^{-9}$	3.46
-	×		.25(	.00635)	Hydrogen	-423 (-252)	3000	.5(.0127)	$1.50 \times 10^{-4}$	148
	×	. •		_					$5.2 \times 10^{-4}$	515
	×					•			$6.1 \times 10^{-4}$	604
	×		•		tog at W				$5.2 \times 10^{-4}$	515
		×							1.8 × 10-4	178
		× >				-			22.0 × 10 <sup>-4</sup>	2178
		< ×	- \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \	- ú		1036 / 667	<b>.</b>	5 ( 0127)	$0.17 \times 10^{-4}$	2 - 1
			.) 67.	(C3 0(3(; ) C7 •	<b>;-</b> -	-423 (-227)	-	(/7.0.) 6.		

TABLE XIX SPECIFIC PERMEABILITY RATES OF DIFFUSION BONDED KAPTON FILM LAMINATES (Continued)

TABLE XIX SPECIFIC PERMEABILITY RATES OF DIFFUSION BONDED KAPTON FILM LAMINATES (Continued)

Specific Permeability 10 <sup>-11</sup> cc-mm/cm²sec.cm Hg	5.70 17.50 12.91 5.78 19.31 29.5 30.29	6.14 15.38 20.19 6.67 25.93 28.51	14.49 2.46 12.71 21.18 7.11 19.99 13.99 23.36	24,15 25,7 19,00 93 27,7 35,6 43,5 55,4
Leak Rate cc/sec	2.88 × 10 <sup>-6</sup> 8.84 × 10 <sup>-6</sup> 6.52 × 10 <sup>-6</sup> 2.92 × 10 <sup>-6</sup> 9.75 × 10 <sup>-6</sup> 14.9 × 10 <sup>-6</sup> 15.30 × 10 <sup>-6</sup> 8.59 × 10 <sup>-6</sup>		1.24 × 10 ° 6 6.42 × 10 ° 6 10.70 × 10 ° 6 3.59 × 10 ° 6 10.10 × 10 ° 6 7.07 × 10 ° 6	12.20 × 10 <sup>-5</sup> 1.30 × 10 <sup>-5</sup> 0.96 × 10 <sup>-5</sup> 4.7 × 10 <sup>-5</sup> 1.40 × 10 <sup>-5</sup> 1.80 × 10 <sup>-5</sup> 2.20 × 10 <sup>-5</sup> 1.50 × 10 <sup>-5</sup> 2.8 × 10 <sup>-5</sup>
Total Laminate Thickness Mils (mm)	1.0 (.0254)	1.0 (.0254)	1.0 (.0254)	1.0 (.0254)
Gold Coating Thigkness A	3000	3000	3000	3000
Test Temperature °F(°C)	+70 (+21)	-320 (-195)	-423 (-252)	-423 (-252)
Permeating Gas	Helium	He i c	Heli ca	Hydrogen
Film Thickness Mils (mm)	.5 (.0127)	.5 (.0127)	.5 (0127)	.5 (.0127)
Film Stress 0% 20%	× × × ×	× × × ×	××××	The state of the s
Concept	Laminated Film (Diffusion Bonded)			

### 3.3.3 Film Flexibility

Twist-flex samples of each laminate material were tested at -252°C to determine their relative flexibilities (see paragraph 3.1.2 for procedures utilized), and the results are tabulated in Table XX . Comparing these results with those shown in Table XVII for Task II, reveals that the twist-flex life of the metallized film laminates, especially the 1/2 mil samples, improved. This increase is attributed to the lower sputtering energy utilized resulting in less film degradation.

## 3.3.4 Bond Strength Determinations

Tables XXI and XXII present the results of the butt- and lapbond tests at -195°C and -252°C. As demonstrated throughout the program a very reliable joint can be made with the diffusion bonded laminates. The failures in this task were mostly in the film itself at stress levels comparable to that obtained with the plain Kapton film.

Configurations of the lap and butt joints are presented in Figure 18

## 3.3.5 Interply Inflation

Interply inflation specimens (see paragraph 3.1.4) were prepared and tested at  $-195^{\circ}$ C and  $-252^{\circ}$ C. No inflation was noted at either temperature for immersion times of 5 and 30 minutes.

TABLE XX - TWIST-FLEX TEST RESULTS @-423°F (-252°C)

TABLE XXI - BOND STRENGTH RESULTS AT -423°F (-252°C)—BUTT JOINT

		Ç.							· · ·												
	Failure	Location		Film	Grib	Film	Film	Film	ü	E .	Film	Film	Grip	Film	נגן	E	Film	Film	FIE	Film	
Joint Stress	Ps:	(Kg/Sq. cm)		25.5 (1.77)	12,2 (0.86)	23.5 (1.65)	21.8 (1.53)	25.7 (1.81)	21 25 /1 40/	(44-1) (7:17	27.0 (1.90)	24.0 (1,69)	14.2 (1.00)	25.0 (1.76)	31 1 (2 10)	(01.2) 1.10	33.3 (2.34)	32.8 (2.31)	33.0 (2.32)	31.8 (2.24)	
Film Stress	Ks:	(Kg/Sq. cm)		51.0 (3585.3)	24.4 (1715.3)	47.0 (3304.1)	43.6 (3065.1)	51,4 (3613.4)	77 50001 20	17. /0/2) 6.24	54.0 (3796.2)	48.0 (3374.4)	28.4 (1996.5)	50.0 (3515.0)	31 1 (2184 3)	(5,0013)	33,3 (2341.2)	32.8 (2306.0)	33.0 (2320.1)	31.8 (2235.8)	
Load		Lbs (Kg)			12,2 (5,53)	23.5 (10,66)	21.8 (9.89)	25,7 (11.66)	21 25 (0 44)	(+0.7) (2.12	27.0 (12.25)	24.0 (10.88)	14.2 (6.44)	25.0 (11.34)	31 1 (14 15)	67.11.10	33.3 (15.10)	32.8 (14.87)	33.0 (14.97)	31.8 (14.42)	
nt	Width	In. (cm)	:	1.00 (2.54)	-	•		►	1 0002 540	(10.1)00.				<b>&gt;</b>	1 00/2 54)	(	-			<b>&gt;</b>	
Joint	Area	Sq.In.(Sq.cm)		1.00 (6.45)				•	1 00 (5 45)	(21.0)				<b>&gt;</b>	1,00 (6,45)	(2: :2)				<b>&gt;</b>	
	Specimen	°Z	,	_	7	ო	4	3		- (	7	က	4	2	•		7	က	4	2	
		Material		Laminated .25 mil	(.00635 mm) Kapton,	3000 A gold			Laminated 25 mil	20000	(.00635 mm) Kapton,	5000 A gold			Laminated .5 mil	7 7 7 7 7 7	( UIZ mm) Kapton,	3000 A gold			_

TABLE XXII- BOND STRENGTH RESULTS AT -423°F (-252°C)—LAP JOINT

		Joint	<del>-</del>	road	Film Stress	Joint Stress	
	Specimen No.	Area Sq.In,(Sq.cm)	Width In. (cm)	Lbs. (Kg)	Ksi (Kg/Sq. cm)	Psi (Kg/Sq.cm)	Failure Location
Laminated , 25 mil		,5 (3.22)	1.00 (6.45)	16.6 (7.53)	33.2 (2334.2)	33.2 (15.06)	Film
,00635 mm) Kapton,	2			16.5 (7.48)	33.0 (2320.1)	33.0 (14.97)	=
3000 Å gold	က			16.6 (7.53)	33.2 (2334.2)	33.2 (15.06)	z
•	4			15.2 (6.89)	30.4 (2123.3)	30.4 (13.78)	=
	2	<b>→</b>	<b>→</b>	11.1 (5.03)	22.2 (1560.8)	22.2 (10.07)	Grip
Laminated .25 mil		. ,5 (3,22)	1,00 (6,45)	19.5 (8.85)	39.0 (2742.0)	39.0 (17.69)	Film
.00635 mm) Kapton,	2			16.4 (7, 44)	32,8 (2249.8)	32,8 (14.87)	=
5000 Å gold	m			20.5 (9.30)	41.0 (2882.6)	41.0 (18.60)	=
	4			0,22 (1,00)	0.44 (30.9)	0.44 (0,20)	Grip
	2	<b>→</b>	<b>→</b>	23.0 (10,43)	46.0 (3234.1)	46.0 (20.86)	Film
Laminated .5 mil	_	.5 (3.22)	1,00 (6,45)	20.8 (9.43)	20,8 (1462.3)	41.6 (18.87)	Film
.0127 mm) Kapton,	2			20.7 (9.38)	20.7 (1455.4)	41.4 (18.78)	Ξ
3000 Å gold	က			19,4 (8,79)	19.4 (1363.9)	38.8 (17.60)	Ξ
	4			19.7 (8.94)	19.7 (1385.0)	39 4 (17.87)	Ξ
	5	<b>→</b>	<b>→</b>	18.0 (8.16)	18.0 (1265.5)	36.0 (16.33)	=

### 4.0 CONCLUSIONS

- a. The objective of this program, to establish the feasibility of reducing the gas permeability of polymesic films, was accomplished. It was demonstrated that the diffusion-bonded laminate concept is capable of reducing the helium or liquid hydrogen permeability of Kapton and Mylar films by 3 to 4 orders of magnitude without drastically altering their flexibility at cryogenic temperatures. Permeability rates as low as  $1 \times 10^{-15}$  cc-mm/cm<sup>2</sup>-sec-cm Hg were obtained. The degree of impermeability obtained with this concept appears more dependent upon the quality of the applied coating than on its thickness. If the porosity of the applied coating is too high, the diffusion bonding process can not statistically seal a significant portion of the holes and a high permeability rate persists.
- b. The diffusion bonded laminate concept is suitable for bladder or component fabrication since it was demonstrated that strong reliable joints can made without affecting the integrity of the film laminate. The joints made with the diffusion bonded laminates were as strong as those made with the plain Kapton film and the type of failure similar.
- c. Metallizing Mylar or Kapton by either vacuum deposition or sputtering reduces the films cryogenic flexibility slightly. There was no evidence however, that the diffusion bonding process itself resulted in further reductions in film flexibility. However, considering the reduction in flexibility due to metallization, the twist-flex life of a Kapton laminate was still equal to that of plain Mylar of equal thickness.
- d. The laminate concept is applicable as a barrier ply for polymeric expulsion devices, cryogenic insulation systems, and as a space-stable bagging material where the combination of low permeability and flexibility is important over a wide temperature range.

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